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HRF Rack One Integration Test Procedure III: Payload Rack Operation Support Procedures

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**HRF Rack One
Integration Test Procedure III:
Payload Rack Operation
Support Procedures**

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ABSTRACT

This document outlines the empty payload rack verification test interface connections between the Human Research Facility Rack with the Payload Rack Checkout Unit (PRCU) individual payload functionals. These procedures facilitate the activities required for the mating and demating of the Human Research Facility (HRF) Rack to the PRCU, execution of offline payload functionals, and the preliminary verification tests for the empty HRF Rack.

The purpose of this procedure is to provide the primary steps necessary for successful integration of the payload rack with the test environment. The payload rack procedure will be conducted in the Building 241 PRCU test environment located at the Johnson Space Center, Houston, Texas. A step-by-step sequence of activities to be conducted is included in Section 6.0 of this document.

A Test Readiness Review (TRR) will be held prior to the start of this activity. The TRR Board, Quality Engineering, and the Payload Test Conductor will agree to proceed with the individual tests listed in this document.

KEY WORDS

Human Research Facility
International Space Station Program

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LIST OF ACRONYMS AND ABBREVIATIONS

ac	Alternating Current
AUX	Auxiliary
BOB	Breakout Box
C&DH	Command & Data Handling
CBL	Commercial Bills of Lading
COM	Communications
COTS	Commercial-Off-The-Shelf
CS	Connector Saver
DAPC	Data Acquisition and Process Controller
DC	Direct Current
DR	Discrepancy Report
DVM	Digital Volt Meter
EXPRESS	EXpedite the PROcessing of Experiments to Space Station
FDS	Flight Data System
FOD	Foreign Object Damage
GSC	Ground Station Control
GSE	Ground Support Equipment
GASMAP	Gas Analyzer System for Metabolic Analysis of Physiology
HRDL	High Rate Data Link
HRF	Human Research Facility
I&ES	Instrument and Experiment Simulator
IPC	Interprocess Communications
IPV	Ion Pump Voltage
IS	Information Systems
ISPR	International Standard Payload Rack
ITCS	Internal Thermal Control System
JSC	Johnson Space Center
LAN	Local Area Network
LCD	Liquid Crystal Display
LED	Light Emitting Diode
LOV	

LIST OF ACRONYMS AND ABBREVIATIONS (CONT'D)

MRDL	Medium Rate Data Link
MS-DOS	Microsoft Disk Operating System
NASA	National Aeronautics and Space Administration
PASS	Primary Avionics Software System
PCMCIA	Portable Computer Memory Card International Adapter
PEPSE	Programmable Electrical Power System Emulator
PRCU	Payload Rack Check-out Unit
QA	Quality Assurance
RAMS	Random Access Mass Spectrometer
RIC	Rack Interface Controller
RS	
SAB	Signal Adapter Box
SIR	Standard Interface Rack
SOL	
SSPCM	Solid State Power Control Module
TCP/IP	Transmission Control Protocol/Internet Protocol
TCS	Thermal Control System
TPS	Task Performance Sheet
TRR	Test Readiness Review
TSS	Test Support System
UIP	Utility Interface Panel
V	Volts
VES	Vacuum Exhaust System
VRDS	Verification Requirements Data Sheet
VRs	Vacuum Resource System
VMM	Volume Measurement Module
VTR	Video Tape Recorder
WS	Workstation

1.0 INTRODUCTION

1.1 PURPOSE

This document provides the procedures necessary to integrate a payload rack into the PRCU test environment, perform offline payload functional checks and execute an interface check between the two systems. The expected end product of this activity is to accomplish the offline payload functionals, successful connection, disconnection, and verification of the empty payload rack. This document does not address any HRF Rack to PRCU instructions other than the initial connection, interface check and final disconnect.

1.2 SCOPE

This document provides task sequencing to satisfy the test requirements as detailed in the document "Rack One HRF Unique Payload Verification Plan" in SSP-57400, "Human Research Facility Unique Payload Verification Plan for Rack 1, International Space Program". The details listed herein describe the necessary hardware (H/W), configuration, test equipment set-ups, instrumentation requirements, data requirements, safety concerns, and all other details necessary to perform the payload integration.

This procedure is applicable to the subsystems and components of the HRF Rack and the PRCU test environment. The initial interface check of the test environment, payload rack, offline payload functionals, and the demating from the PRCU are described herein.

A notation of To Be Determined, (TBD) has been included throughout this document to signify subsystems that are to be incorporated into the PRCU test environment and/or HRF Rack, but are not functional at this time. Upon completion of the subsystems, the hardware will undergo acceptance testing. Subsystem procedure updates will be included in subsequent document revisions.

1.3 DOCUMENT OVERVIEW

This document details the test setup, tear down, and procedures divided into five (5) sections:

- Section 6.1 MATING OF PAYLOAD RACK TO INTERNATIONAL STANDARD PAYLOAD (ISPR) PANEL
- Section 6.2 EMPTY RACK TESTS
- Section 6.3 OFFLINE PAYLOAD FUNCTIONALS
- Section 6.4 DEMATING OF PAYLOAD RACK FROM ISPR PANEL
- Section 6.5 LAN INTERFACES

1.3.1 Document Hand-Write Change Control

This document is designed to present baseline procedures for pre-integration testing of the payload and HRF Rack. It is therefore assumed that this document is subject to hand-write changes while in use in the test area. Hand-write entries will be controlled and documented in this procedure. All hand-writes must be approved by Quality Engineering and the Test Conductor prior to implementation. Quality Assurance will validate all hand-writes. If safety is affected, then Safety Personnel must also approve changes. The personnel that have Task Performance Sheet (TPS) signature authority are authorized to make hand-write changes to this document. Hand-written changes to this document will be done using deviation sheets (See Appendix A). This document will be revised to include permanent hand written changes.

1.3.2 Warnings and Cautions

Prior to performing any operation, test personnel must be familiar with all "General Notes, Warnings, Cautions, Special Instructions and Safety Precautions" contained in the reference documents and drawings unless otherwise specified within this procedure.

1.3.3 Task Sequencing

The procedures outlined in this document are written to assure technical completion of a specified task and are not necessarily sequenced to provide optimum crew/tool equipment utilization or rack build up. The work is to be accomplished sequentially, unless it is more efficient to parallel the operations or the secondary document procedures. The responsible Test Conductor must first evaluate any change to assure that there is no degradation of technical requirements, system safety, personnel safety, scheduling, etc. Sequencing changes require concurrence from Quality Assurance.

1.3.4 Repeat Operations

Prior to proceeding, operations that must be repeated require approval of the Test Conductor, and Quality Assurance. All repetitive operations must be documented in the Repetitive Operations Log in Appendix A.

1.3.5 Discrepancies

If any discrepancy occurs in the form of an equipment failure, hazard, or emergency, the personnel concerned will take appropriate action to ensure personnel and equipment safety, and report to a Quality Assurance Specialist. The Test Conductor will notify the National Aeronautics and

Space Administration (NASA) facility manager and act as focal point for any further effort required. If required, a Discrepancy Report (DR), Johnson Space Center (JSC) form 2176 will be initiated by Quality Assurance, and will be tracked and worked as described in document NT1-ADM-013 (See Appendix A).

1.3.6 Safety Support

JSC Safety, Health Requirements established in document JPG 1700.1 Version H, will be strictly adhered to throughout all phases of test activities. All hazardous activities will be coordinated with the appropriate facility personnel.

1.3.7 Emergency/Accident Procedure

The following procedures are to be used in the event of an emergency situation, (i.e. smoke or fire) or in the case of an accident involving personal injury.

Emergency procedures provide pre-planned and approved guidelines for handling potential hardware/software malfunctions and hazardous situations. If a hazardous situation occurs, the following definitions state the actions necessary to maintain control of the situation and personnel safety. Actions required for the situations not covered by these procedures shall be provided by the Test Conductor real-time, based on his/her best judgment.

Definitions

Abort Test: Take immediate and rapid actions for restoration of safe conditions removal or rescue of test personnel, notification of the appropriate personnel about the hazardous situation, and shutdown of all systems. This action is taken in catastrophic critical hazard conditions such as fire, smoke, or serious personnel injuries.

Terminate Test: Discontinue test per the standard shutdown procedures provided. This action is required when the situation prevents further compliance with the test objectives.

Hold and Evaluate: Maintain current test conditions or proceed to safe mode to allow time to review system status and impacts of the situation. This action is required in the event of a hardware/software malfunction.

Emergency Exits and Equipment

Figure 1-1 shows the emergency exits for personnel in the test area; and shows the location of fire pull-stations and fire extinguishers. Figure 1-2 shows the emergency meeting place outside of Building 241.

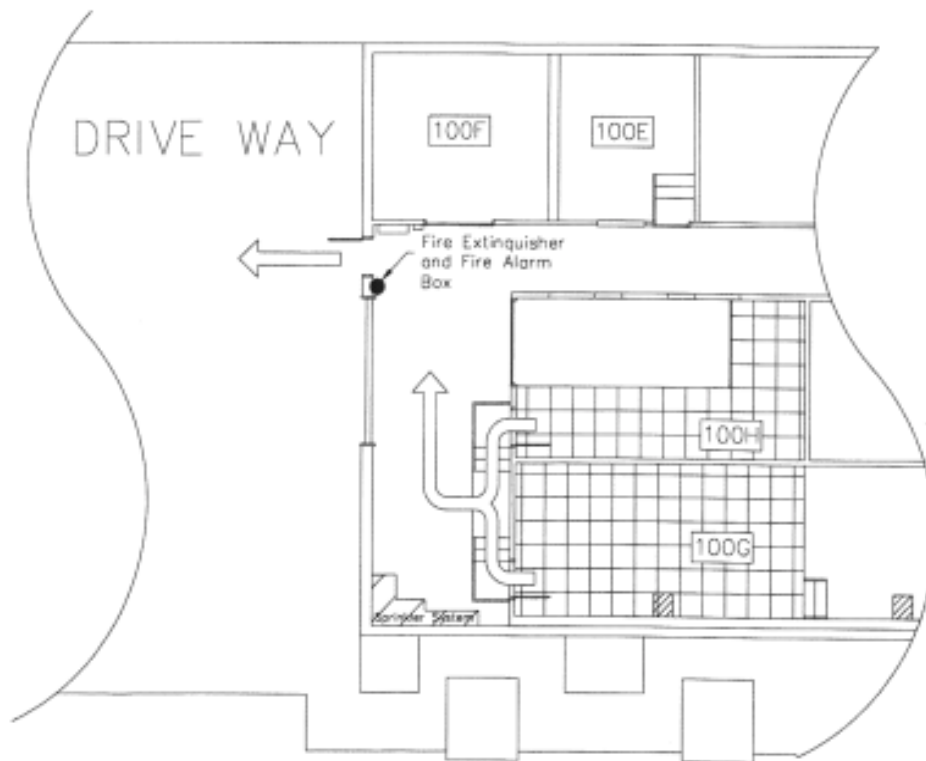


Figure 1-1 241 Facility Clean Room Emergency Exits

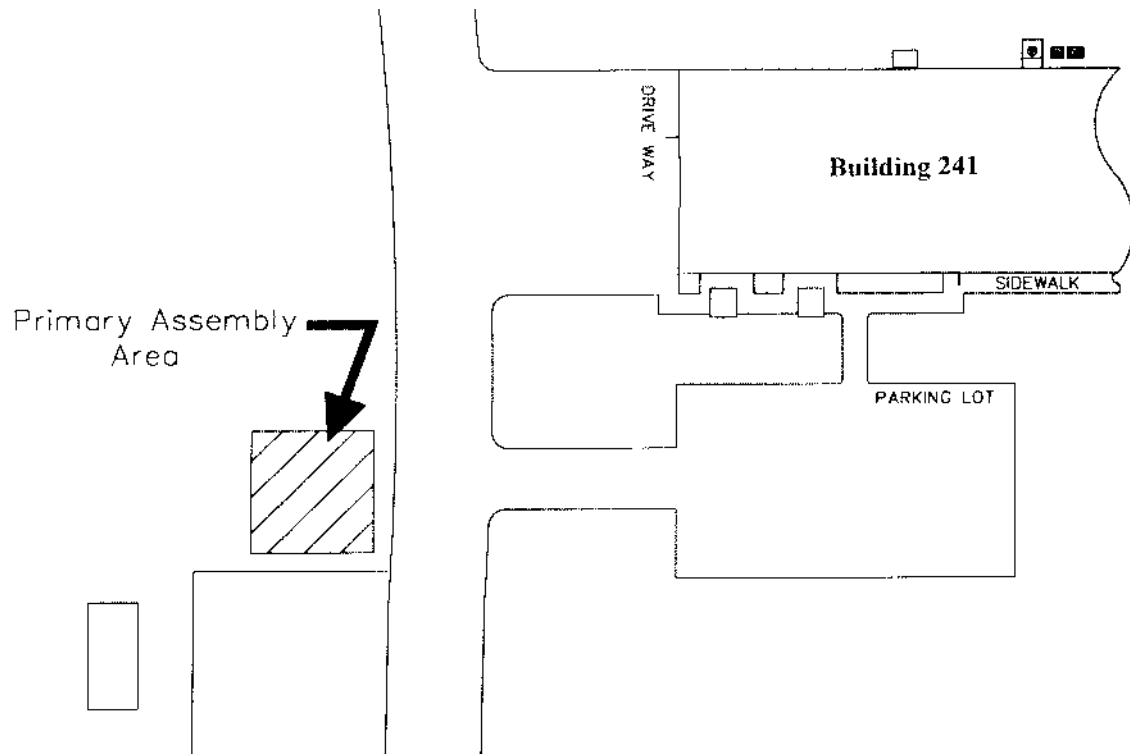


Figure 1-2 241 Facility Emergency Meeting Place

Emergency/Accident Reporting

The Facility Engineer has the primary responsibility of initiating the notification process. General Emergency Instructions:

- (1) Sound the alarm and evacuate the area.
- (2) If safe, render/de-energize energy systems.
- (3) Initiate Flash reporting sequence.
- (4) Establish emergency response team to support follow on action.

Figure 1-3 shows the JSC Emergency Number and Reporting Sequence. This number is a coordinated number for the emergency related medical, fire and security groups at JSC.

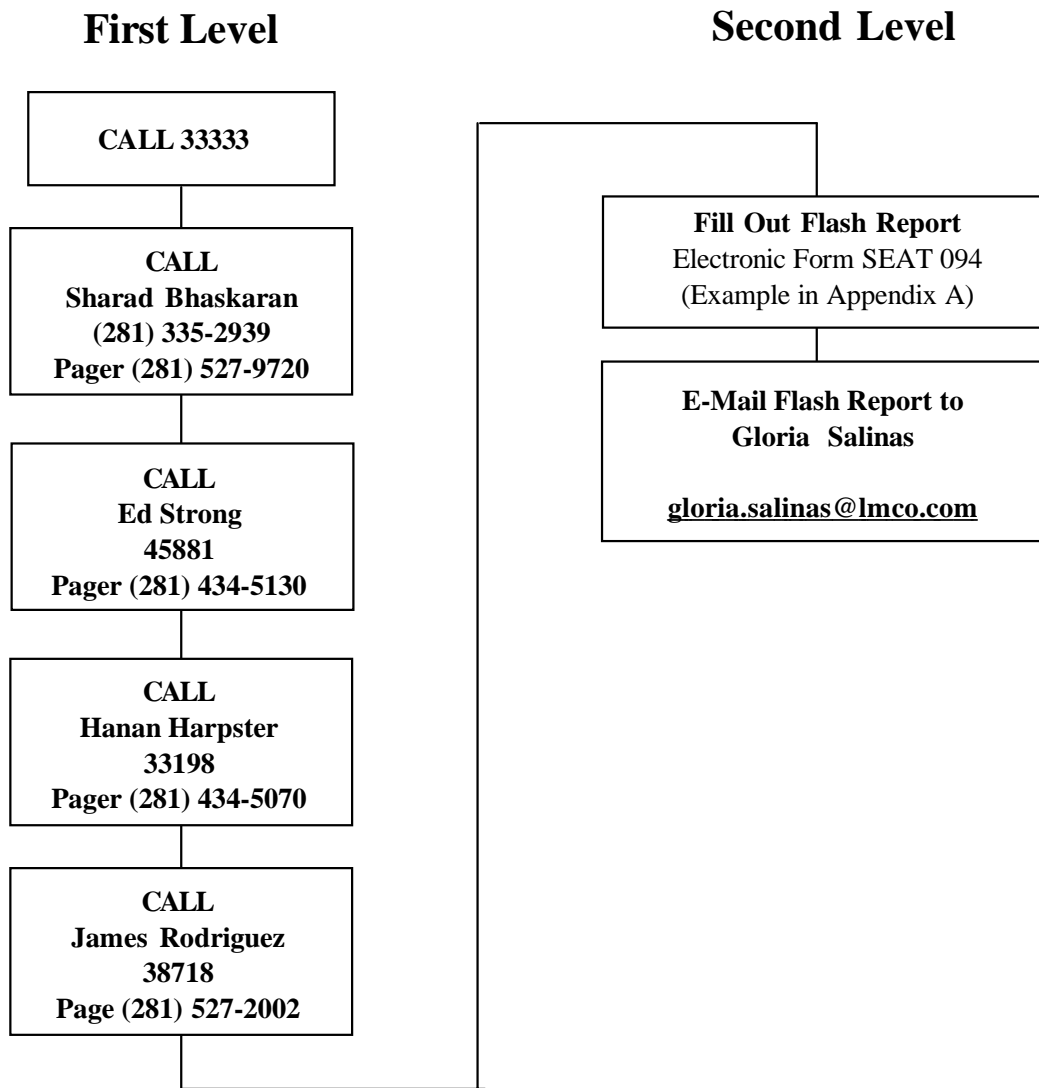


Figure 1-3 JSC Emergency Number and Reporting Sequence

Systems Emergency Procedures

The following procedures are to be carried out by the Test Conductor and Test Personnel in accordance with the condition as defined below:

CONDITION	RESPONSIBILITY	ACTION
Fire/Visible Smoke in Test Area	Test Conductor/Technician	Abort Test

- (1) Sound the alarm: Activate alarm at pull box and/or phone in emergency.
- (2) Do not move injured personnel unless necessary to prevent further injury.
- (3) If safe, attempt to de-energize system, i.e. thermal, electric, etc.
- (4) Initiate notification process. This may be conducted away from the situation from a telephone.

CONDITION	RESPONSIBILITY	ACTION
Electrical burn/smoke odor	Test Conductor/Technician	Terminate Test

- (1) Shutdown all electrical test equipment systems.
- (2) Locate nearest fire extinguisher.
- (3) Investigate/Isolate the source of odor.
- (4) If required, perform steps associated with a Fire/Smoke situation.

CONDITION	RESPONSIBILITY	ACTION
Loss of Facility Power	Test Conductor/Technician	Hold & Evaluate

- (1) Evaluate the situation and impact to test. Investigate the cause and potential frequency of occurrence. Take appropriate steps to restore the failed systems to their nominal/safe operating conditions.

Personnel Emergency/Accident Procedures

CONDITION	RESPONSIBILITY	ACTION
Serious Personal Injury	Test Conductor/Technician	Terminate Test

- (1) To prevent further injury, do not move the injured personnel unless necessary.
- (2) Render the area safe, then administer first aid as required.
- (3) Initiate notification process.

Do not leave injured personnel alone until emergency personnel arrive.

CONDITION	RESPONSIBILITY	ACTION
Minor Personal Injury	Test Conductor/Technician	Hold & Evaluate

- (1) Render the area safe, then administer First Aid as required.
- (2) Initiate notification process.
- (3) Take injured individual to medical treatment facility.

1.3.8 Hazardous Waste Handling

Hazardous material identification, labeling and storage at Building 241 shall be done according to JSC Form 1161, "Disposal Inventory for Miscellaneous Hazardous Wastes." Disposal containers, transportation and disposal will be provided by the designated JSC waste management service. All Internal Thermal Control System (ITCS) waste disposal in Building 241 should be coordinated through the Facility Manager.

2.0 APPLICABLE DOCUMENTATION

The following documents form a part of this Verification Plan to the extent specified. Tasks and activities referenced in pre-test, post-test, and procedural sequences may be performed using the most recent revision of the document stated.

NASA Documents:

Number	Rev.	Title
JHB 5322	C	Contamination Control Requirements Manual
KHB 1700.7	LI	Space Shuttle Payload Group Safety Handbook
LS-71135-3	A	Human Research Facility Integration Flight Prototype Rack Interface Verification Test
NT1-ADM-012	Base-line	Task Performance Sheet (TPS) NT/Occupational Safety and Institutional Assurance Division
NT1-ADM-013	A	Quality Assurance Record Center Discrepancy Reporting and Tracking Systems
SSP57400		Human Research Facility Unique Payload Verification Plan for Rack 1, International Space Program
LS-71139-2	B	HRF Flight Rack One Integration Test Procedure II: Payload Rack Checkout Unit Mechanical Operations and Fluid Sampling
LS-71139-5	B	HRF Flight Rack One Integration Test Procedure V: Rack Activation/Deactivation

Boeing Documents:

Number	Rev.	Title
D683-44094-2	A	Human Research Facility Flight Rack Command & Data Handling (C&DH) Acceptance Test Procedure
D683-27519-1	G	User Guide for the Payload Rack Checkout Unit (PRCU)

2.1 APPLICABLE SOFTWARE

The following software provides the configuration data used in this test setup:

Payload HRF Rack Configurations are based upon:

Software Item	Version
Rack Interface Controller (RIC)	Expedite the Processing of Experiments to Space Station (EXPRESS) -9
EXPRESS Laptop	EXPRESS HH

PRCU Software Configurations are based upon:

Software Item	Version
Payload Rack Check-out Unit (PRCU)	PRCU Block 2.0
Payload Executive Processor (PEP)	PEP Version 18

HRF Software Configurations are based upon:

Software Item	Version
Common Software	Block 2.3
Commercial-Off-The-Shelf (COTS) Applications	N/A
Instrument Applications	N/A
Experiment Applications	N/A
Workstation software	Load 5.0
Ultrasound Software	Rev 108.17
Gas Analyzer System for Metabolic Analysis of Physiology (GASMAP) Software	I/F Shell: 1AX20 Random Access Mass Spectrometer (RAMS): 1AX60

3.0 TESTING PROCESS OVERVIEW

3.1 TESTING OBJECTIVE

The test objectives are as follows:

- Facilitate the successful connection of the HRF Rack with the PRCU ISPR panel.
- Perform off-line functional checks prior to the installation of the payload into the HRF Rack.
- Perform functional checkout tests of the empty Payload HRF Rack.
- Facilitate successful disconnection of the HRF Rack to the PRCU ISPR panel.

3.2 TEST REQUIREMENTS

The following paragraphs describe the requirements of the specific tests to be conducted and may include references to the specific Verification Requirements Data Sheet (VRDS) to be completed.

3.3 TEST CONDITIONS

3.3.1 Test Conduct Ground Rules

The rules as defined in the following subparagraphs will be followed during all test activities.

3.3.2 Roles and Responsibilities

The Test Conductor is responsible for the overall management and integration of all verification testing at the systems level. The Test Conductor is responsible for the safe, successful control and conduct of all testing. The Test Conductor will assure all test team members are knowledgeable of the subsystems required for the verification test to be performed. The conductor acquires and assigns test resources and is responsible for the adequacy of test documentation. Additional responsibilities are:

- Test schedule coordination
- Test resource management
- Assurance of efficient test conduct
- Data and reports coordination

The Test Engineer is responsible for conducting the specific verification testing, including the coordination of test materials and personnel. The Test Engineer provides the test configuration, test plan and required

paperwork/procedures. The Test Engineer is the principal technical focal point for a given test. The Test Engineer coordinates all test data processing and supports the Test Conductor in the preparation of the post test report.

The Facility Engineer is the member responsible for ensuring that the required instrumentation is calibrated, installed and conditioned to provide the data necessary to meet the test objectives. The Facility Engineer is responsible for the coordination of certified Test Technician/Test Operator support.

The Test Technician/Test Operator is responsible for selection, setup, operation, maintenance and configuration of the test equipment in accordance with the approved test plan and procedure.

3.3.2.1 Test Area Requirements

Special emphasis is to be given to testing flight articles. The following rules will be incorporated into test documentation and compliance is the responsibility of all test team members. Repeated non-compliance may be grounds for denial of access to the test facility.

3.3.2.2 Test Area Cleanliness

Room 100H in Building 241 is certified as a level 100K clean room. Requirements for working in such an environment are detailed in Contamination Control document, JHB 5322C. All test team members with access to room 100H shall be familiar with these requirements and may undergo pre-access training or certification at the discretion of the Facility Engineer. The following rules shall be maintained at all times while in the test facility:

- Smocks, head and beard covers, shall be worn at all times.
- Test Area will be kept clean and orderly at all times.
- All debris created during test preparation, conduct, or tear down will be continuously removed to prevent Foreign Object Damage (FOD) contamination.

3.3.2.3 Test Area Access

Access to all test areas shall be limited during test operations. Only essential personnel shall be admitted. The test area, surrounding test consoles, and test instrumentation shall be controlled to restrain visitors and prevent tampering with the test article or test equipment. Determination of essential personnel will be made by the Test Conductor or Test Engineer, and enforced by the Facility Engineer.

3.3.2.4 Work Area Rules

The following work rules shall be observed for the duration of testing:

- All work stands shall have toe boards sufficient to prevent any item from being accidentally dropped into a test article.
- All work stands shall have the side accessing the test article padded to prevent test article damage in the event the stand comes in contact with the test article.
- Rings and watches must be taped or removed.

3.3.2.5 Temporary Configuration Changes

Temporary changes to the Test Article configuration will be accomplished and documented as described in document NT1-ADM-012 TPS NT/Occupational Safety and Institutional Assurance Division.

4.0 TPS AUTHORIZED PERSONNEL

The TPS Authorization is comprised of two (2) types:

- Type A – A Task Performance Sheet that changes the temporary or permanent configuration of the “Flight” (Class I) or Ground Support Equipment (GSE) test hardware. These documents must be reviewed and agreed upon by the customer before obtaining a NASA Signature. Absolutely no work is to be performed without having the proper paperwork in hand with the appropriate signatures.
- Type B – A Task Performance Sheet that does not change the configuration of the hardware which is being tested. These documents, do not require a NASA Signature, and are to be coordinated with the customer and submitted for signature.

All documents must have the signature of the Lockheed Martin engineer authority in charge of verification.

If documents require hardware to be pulled out of bond; the appropriate signature authority for the bond room must be included. This list is for reference purposes only, verify before use. The official list is provided in NASA EA 5 memo.

LIST OF AUTHORIZED SIGNATURES

Project ID	Project Name	New Project ID	New Project Name	NASA Technical Monitor	Mission Assigned	Other Authorized Signatures
HPMHPMS1	Integration Hardware Definition & Development/Ground Rack Design and Build	HPMS	High Fidelity Mockup/Ground Development Facility/Launch Integration Facility/Payload Rack Checkout Unit	Ed Strong	HRF	Sharad Bhaskaran Robert Henneke Bob Trittipio Tom Wiggins Elton Witt
HPM1	Ground Facilities Development	Deleted – Content moved to HPMS				
HPM3	Water Cooled Rack Development	HPM3	Flight Prototype Rack Integration/Flight Rack Integration	Ed Strong	HRF	Carlos Aquilar Sharad Bhaskaran Todd Leger Kevin Upham
HPCP	HRF Launch Package 1 Hardware Design	Deleted – Content moved to HPM3				
MEIT	Multiple Element Integration Test (MEIT)	Deleted – Content moved to HPM3				

5.0 TEST SET UP

The test set-up and tear down will be governed by Task Performance Sheets (TPS) JSC form 1225.

5.1 PRE-TEST ACTIVITY

N/A

5.2 POST-TEST ACTIVITY

N/A

6.0 TEST PROCEDURE

The following procedure contains steps to mechanically connect/disconnect the HRF Rack to/from the PRCU test environment, perform an interface check between the HRF Rack and PRCU, and to execute individual functional checks of payload drawers. The PRCU provides a testing environment that will aid in the closure of verifications of the empty HRF Rack.

6.1 MATING OF PAYLOAD RACK TO INTERNATIONAL STANDARD PAYLOAD RACK (ISPR) PANEL

The following sections detail the interface connections between the HRF Rack and the PRCU test environment. Several connections can be made at the discretion of the test conductor. These connections are dependent upon the test to be performed including: Flight Data System (FDS) Maintenance, Vacuum and GN₂.

NOTE: The Utility Area Close-out T-Bar Assy may remain in configuration or be removed to prevent interference with future activities, per the discretion of the test conductor.

6.1.1 Pre-Mate Checks

Before connecting any hardware, an inspection must occur to verify that no damage has occurred to the connector. If a connector is damaged, the hardware cannot be connected. A DR must be written against the cable and the discrepant connector. The part must be replaced, or the hardware repaired. The DR is dispositioned before proceeding on with connecting the HRF Rack to the PRCU test environment.

TABLE 6.1 PRE-MATE CHECKS

Step	PRCU	HRF Rack 1	HRF Rack 2										
1.	Verify the following: <ul style="list-style-type: none">• PRCU fluid sample is in spec per SSP30573• Programmable Electrical Power System Emulator (PEPSE) facility breaker is OFF												
2.	Inspect the following hardware for damage and record serial numbers: <table><tr><th>Item</th><th>P/N</th></tr><tr><td>Main Pwr Connector Saver Assy</td><td>683-27524-1</td></tr><tr><td>Auxiliary (AUX) Pwr Connector Saver Assy</td><td>683-27524-2</td></tr><tr><td>1553B Bus A Con-Connector Saver Assy</td><td>683-27524-3</td></tr><tr><td>1553B Bus B Con-Connector Saver Assy</td><td>683-27524-4</td></tr></table>	Item	P/N	Main Pwr Connector Saver Assy	683-27524-1	Auxiliary (AUX) Pwr Connector Saver Assy	683-27524-2	1553B Bus A Con-Connector Saver Assy	683-27524-3	1553B Bus B Con-Connector Saver Assy	683-27524-4		
Item	P/N												
Main Pwr Connector Saver Assy	683-27524-1												
Auxiliary (AUX) Pwr Connector Saver Assy	683-27524-2												
1553B Bus A Con-Connector Saver Assy	683-27524-3												
1553B Bus B Con-Connector Saver Assy	683-27524-4												

TABLE 6.1 PRE-MATE CHECKS (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
	High Rate Data Link (HRDL) Connector Saver Assy 683-27524-5 Video Connector Saver Assy 683-27524-6		
3.	Inspect the following hardware for damage and record serial numbers: <u>Item</u> <u>P/N</u> FDS Connector Saver Assy 683-27524-7 Local Area Network (LAN) 1 Connector Saver Assy 683-27524-8 LAN 2 Connector Saver Assy 683-27524-9 Video Connector Saver Assy 683-27524-10 1553 Address Terminator SEG38116104-301		
4.	Inspect the following hardware for damage and record serial numbers: Mod Thermal Control System (TCS) Supply Connector Saver Assy 683-27554-2 Mod TCS Return Connector Saver Assy 683-27554-4		N/A:_____ T:_____ QA:_____
5.	Inspect the following hardware for damage and record serial numbers: Mod TCS Supply 683-27554-2 GSE Transfer Supply Hose SKL83-45519-3 GSE Transfer Return Hose SKL83-45519-2		N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.2 Moderate Temperature Cooling Interface

There are two (2) possible methods for mating the HRF Rack to the PRCU Moderate Temperature Cooling System. The PRCU connector saver assemblies, or the GSE Transfer Hoses and Accumulator can be utilized. The test conductor determines which method should be used. System operating procedures are dependant upon the method used and must be developed accordingly. The method used will be noted on the TPS.

6.1.2.1 PRCU Connector Savers

The PRCU thermal connector savers are for use with the HRF Rack and do not contain any valves. A self-sealing Quick Disconnect (QD) Connector is located at each end of the hose. Once mated, the only valves for controlling thermal flow are the automated valves in the test environment and the flight rack.

TABLE 6.2 PRCU CONNECTOR SAVERS

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Disconnect the following hardware: • Thermal Control System (TCS) Mod Supply hose from T-Bar Assy	
2.		Inspect the following hardware for damage: • TCS Mod Supply hose • Connector Saver (CS) Assy 683-27554-2	
3.		Connect the following hardware: • TCS Mod Supply hose P/N to CS Assy P/N 683-27554-2. Remove Cap cover from TCS Mod Hose by turning and pulling to remove from the bottom	
4.	Connect the following hardware: • CS Assy P/N 683-27554-2 to ISPR Panel (MOD TEMP TCS SUPPLY)		
5.		Disconnect the following hardware: • TCS Mod Return hose from T-bar Assy	
6.		Inspect the following hardware for damage: • TCS Mod Return hose • CS Assy 683-27524-4	
7.		Connect the following hardware: • TCS Mod Return hose to CS Assy P/N 683-27524-4	
8.	Connect the following hardware: • CS Assy P/N 683-27524-4 to ISPR Panel (MOD TEMP TCS RETURN)		
9.	Visually inspect hoses for leaks.		
10.		Connect the following hardware: • TCS Mod Supply hose to T-bar Assy • TCS Mod Return hose to T-bar Assy	N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.2.2 GSE Transfer Hose

The GSE transfer hoses contain manual inlet and outlet valves. The return hose contains an accumulator that offsets the changes in pressure when the rack is in transit.

TABLE 6.3 GSE TRANSFER HOSE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Verify the following: • GSE Transfer Supply • Hose Valve - closed • GSE Transfer Return • Hose Valve - closed	

TABLE 6.3 GSE TRANSFER HOSE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
2.		Disconnect the following hardware: • TCS Mod Supply hose from T-bar Assy	
3.		Inspect the following hardware for damage: • TCS Mod Supply hose • GSE Transfer Supply Hose	
4.		Connect the following hardware: • TCS Mod Supply hose is mated to GSE Transfer Supply hose P/N SK683-45519-3	
5.	Connect the following hardware: • GSE Transfer Supply hose P/N SK683-45519-3 to ISPR Panel (MOD TEMP TCS SUPPLY)		
6.		Disconnect the following hardware: • TCS Mod Return hose from T-bar Assy	
7.		Inspect the following hardware for damage: • TCS Mod Return hose P/N • GSE Transfer Return Hose	
8.		Verify the following: • TCS Mod Return hose is mated to GSE Transfer Return hose P/N 683-27554-4 • GSE Transfer Return hose	
9.	Connect the following hardware: • GSE Transfer Return hose SK683-45519-2 to ISPR Panel (MOD TEMP TCS RETURN)		
10.	Visually inspect hoses for leaks.		
11.		Connect the following hardware: • TCS Mod Supply hose to T-bar Assy • TCS Mod Return hose to T-bar Assy	N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.3 GN₂ Interface

The GN₂ interface provides the HRF Rack with a supply of gaseous nitrogen.

TABLE 6.4 GN₂ INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Disconnect the following hardware: • GN ₂ hose from T-bar Assy	
2.		Inspect the following hardware for damage: • GN ₂ hose • CS Assy 683-27587-1	
3.		Connect the following hardware: • GN ₂ hose P/N to CS Assy P/N 683-27587-1	
4.	Connect the following hardware: • CS Assy P/N 683-27587-1 to ISPR Panel (GN ₂)		
5.		Connect the following hardware: • GN ₂ hose to T-bar Assy	N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.4 Vacuum Interface

TABLE 6.5 VACUUM INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Disconnect the following hardware: • Vacuum Resource hose from T-bar Assy	
2.		Inspect the following hardware for damage: • Vacuum Resource hose • CS Assy P/N 683-27552-1	
3.		Connect the following hardware: • Vacuum Resource hose • CS Assy P/N 683-27552-1	
4.	Connect the following hardware: • CS Assy P/N 683-27552-1 to ISPR Panel Vacuum Resource System (VRS)		
5.		Disconnect the following hardware: • Vacuum Waste hose from T-bar Assy	
6.		Inspect the following hardware for damage: • Vacuum Waste hose • CS Assy P/N 683-27552-2	
7.		Connect the following hardware: • Vacuum Waste hose to CS Assy P/N 683-27552-2	
8.	Connect the following hardware: • CS Assy P/N 683-27552-2 to ISPR Panel Vacuum Exhaust System (VES)		
9.		Connect the following hardware: • Vacuum Resource hose to T-bar Assy • Vacuum Waste hose to T-bar Assy	N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.5 Power Interfaces

6.1.5.1 Main Power Interface

TABLE 6.6 MAIN POWER INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: • POWER (J1) Light Emitting Diode (LED) is off • AUX POWER (J2) LED is off		
2.	Disconnect the following hardware: • Dust cap from ISPR panel (J1)		
3.		Disconnect the following hardware: • Utility Interface Panel (UIP)-P1 Cable from HRF Rack (Dust cap J1)	
4.		Inspect the following hardware for damage: • UIP-P1 Cable • CS Assy (J1) P/N 683-27524-1	

TABLE 6.6 MAIN POWER INTERFACE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
5.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P1 to CS Assy (J1) P/N 683-27524-1 • CS Assy (E1) P/N 683-27524-1 to Rack Handler Base P/N 220G07455-001 	
6.	Connect the following hardware: <ul style="list-style-type: none"> • CS Assy (P1) P/N 683-27524-1 to ISPR Panel (J1 POWER) 		

T:_____ QA:_____

6.1.5.2 Auxiliary Power Interface

TABLE 6.7 AUXILIARY POWER INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • POWER (J1) LED - off • AUX POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> • Dust cap from ISPR panel (J2) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P2 Cable from HRF Rack (Dust cap J2) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P2 Cable • CS Assy P/N 683-27524-2 	
5.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P2 to CS Assy (J2) P/N 683-27524-2 • CS Assy (E1) P/N 683-27524-2 to HRF Rack Handler Base P/N 220G07455-001 	
6.	Connect the following hardware: <ul style="list-style-type: none"> • CS Assy (P2) P/N 683-27524-2 to ISPR Panel (J2 AUX POWER) 		

T:_____ QA:_____

6.1.6 Video Interface

TABLE 6.8 VIDEO INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • Main Power (J1) LED - off • AUX Power (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> • Dust cap from ISPR panel (J16) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P16 Cable from HRF Rack (Dust cap J16) 	

TABLE 6.8 VIDEO INTERFACE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P16 Cable • CS Assy P/N 683-27524-6 	
5.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P16 to CS Assy (J16) P/N 683-27524-6 	
6.	Connect the following hardware: <ul style="list-style-type: none"> • CS Assy (P16) P/N 683-27524-6 to ISPR Panel (J16 VIDEO, LINE 1) • CS Assy (P16) P/N 683-27524-6 to ISPR Panel (J16 VIDEO, LINE 2) • CS Assy (P16) P/N 683-27524-6 to ISPR Panel (J16 VIDEO, SYNC) 		
7.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P77 Cable from HRF Rack (Dust cap J77) P/N SEG46117303-301 	N/A:_____ T:_____ QA:_____
8.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P77 Cable • CS Assy P/N 683-27524-10 	N/A:_____ T:_____ QA:_____
9.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P77 to CS Assy (J77) P/N 683-27524-10 	N/A:_____ T:_____ QA:_____
10.	Connect the following hardware: <ul style="list-style-type: none"> • CS Assy (P77) P/N 683-27524-10 to ISPR Panel (J77 VIDEO) 		N/A:_____ T:_____ QA:_____

T:_____ QA:_____

6.1.7 HRDL Interface

TABLE 6.9 HRDL INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • POWER (J1) LED - off • AUX POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> • Dust cap from ISPR panel (J7) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P7 Cable from HRF Rack (Dust cap J7) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P7 Cable • CS Assy P/N 683-27524-5 	
5.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P7 to CS Assy (J7) P/N 683-27524-5 	
6.	Connect the following hardware: <ul style="list-style-type: none"> • CS Assy (P1) P/N 683-27524-5 to ISPR Panel (J7-TX HRDL) 		

T:_____ QA:_____

6.1.8 FDS Maintenance Interface

TABLE 6.10 FDS MAINTENANCE INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off AUX POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> Dust cap from ISPR panel (J43) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P43 Cable from HRF Rack (Dust cap J43) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P43 Cable CS Assy P/N 683-27524-7 	
5.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P43 to CS Assy (J43) P/N 683-27524-7 	
6.	Connect the following hardware: <ul style="list-style-type: none"> CS Assy (P43) P/N 683-27524-7 to ISPR Panel (J43 FDS Maintenance) 		

T:_____ QA:_____

6.1.9 1553B Interfaces

6.1.9.1 Bus A Interface

TABLE 6.11 BUS A INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> Dust cap from ISPR panel (J3) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P3 Cable from HRF Rack (Dust cap J3) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P3 Cable CS Assy P/N 683-27524-3 	
5.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P3 to CS Assy (J3) P/N 683-27524-3 CS Assy (P7) P/N 683-27524-3 to 1553 ADDRESS TERMINATOR Assy (J7) P/N SEG38116104-301 S/N 1001 	
6.	Connect the following hardware: <ul style="list-style-type: none"> CS Assy (P1) P/N 683-27524-3 to ISPR Panel (J3 BUS A) 		

T:_____ QA:_____

6.1.9.2 Bus B Interface

TABLE 6.12 BUS B INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> Dust cap from ISPR panel (J4) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P4 Cable from HRF Rack (Dust cap J4) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P4 Cable CS Assy P/N 683-27524-4 	
5.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P4 to CS Assy (J4) P/N 683-27524-4 	
6.	Connect the following hardware: <ul style="list-style-type: none"> CS Assy (P1) P/N 683-27524-4 to ISPR Panel (J4 BUS B) 		

T:_____ QA:_____

6.1.10 LAN Interfaces

6.1.10.1 LAN 1 Interface

TABLE 6.13 LAN 1 INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> Dust cap from ISPR panel (J46) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P46 Cable from HRF Rack (Dust cap J46) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P46 Cable 	
5.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P46 to CS Assy (J46) P/N 683-27524-8 	
6.	Connect the following hardware: <ul style="list-style-type: none"> CS Assy (P1) P/N 683-27524-8 to ISPR Panel (J46 LAN 1) 		

T:_____ QA:_____

6.1.10.2 LAN 2 Interface

TABLE 6.14 LAN 2 INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off POWER (J2) LED - off 		
2.	Disconnect the following hardware: <ul style="list-style-type: none"> Dust cap from ISPR panel (J47) 		
3.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P47 Cable from HRF Rack (Dust cap J47) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P47 Cable CS Assy P/N 683-27524-9 	
5.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P47 to CS Assy (J47) P/N 683-27524-9 	
6.	Connect the following hardware: <ul style="list-style-type: none"> CS Assy (P1) P/N 683-27524-9 to ISPR Panel (J47 LAN 2) 		

T:_____ QA:_____

6.2 EMPTY RACK TESTS

6.2.1 Power

This section details the procedures to perform the power test on the empty HRF Rack. This test verifies the electrical power system of the HRF Rack. The purpose of this test is to ensure that the HRF Rack meets the VRDS criteria.

This test collects baseline power data for the Human Research Facility (HRF) Rack. There is no pass/fail criteria associated with any data collection. Test results are used to determine operational parameters during testing in the specific environment. This test assumes that the test environment can provide sufficient simulation of the ISS Medium Temperature Cooling Loop (MTCL) and power systems.

The system flow sensors are monitored and compared to HRF Rack internal sensors. The test environment power system voltage output is set and verified to 116vdc, 120vdc, and 126vdc demonstrating nominal rack operation.

After twenty (20) minutes, following power application to the HRF Rack, the following steps may be performed using the Primary Avionics Software System (PASS-1000 or the equivalent to collect Health and Status data.)

Once the Main Power sequence is completed, repeat using Auxiliary Power. Test results will be recorded on the Health and Status Data Sheet (Appendix A, Figure 10).

TABLE 6.15 POWER

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Verify the following mates: <ul style="list-style-type: none"> • HRF Rack to PRCU ISPR Panel X • HRF Rack to EXPRESS Laptop NOTE: This step can be used on all three (3) ISPR channels (X=1,2, or 3).	
2.			Digital Volt Meter Verify the following: Main pwr sw - off Record the following information: Model: _____ NASA Tag #: _____ Calibration #: _____ Calibration Date: _____
3.	Connect the following hardware: <ul style="list-style-type: none"> • Digital Volt Meter to ISPR Panel X (TP1) NOTE: This step can be used on all three (3) ISPR channels (X=1,2, or 3).		
4.	Activate PEPSE Rack per document LS-71139-2, Section 6.0		
5.	Activate Computational Rack per document LS-71139-2, Section 6.0		
6.	Activate Data Acquisition and Process Controller (DAPC) Rack per document LS-71139-2, Section 6.0		
7.	Activate Moderate Temperature Chiller per document LS-71139-2, Section 6.0		
8.		Activate PASS-1000 per document LS-71139-2, Section 6.0	
9.		Initiate Command and Data Handling per document LS71139-2, Section 6.0	
10.		HRF Rack, Front Panel Verify the following: <ul style="list-style-type: none"> • "Maintenance" sw - on 	
11.			Digital Volt Meter Main pwr sw - on
12.	PEPSE Rack, Front panel Channel X Main: Select: "ARM" button pwr sw - on NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
13.		ISPR Panel X Verify the following: <ul style="list-style-type: none"> • Power Indicator is orange NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	

TABLE 6.15 POWER (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
14.	PEPSE Rack, Front Panel Channel X Main Record the following: • Voltage: _____ V dc NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
15.		ISPR Panel X, Using Digital Volt Meter (DVM) Record the following: • TP1 Voltage: _____ V dc NOTE: This step can be used on all ISPR Channels (X=1, 2, or 3).	
16.			Activate HRF Rack, EXPRESS Laptop per LS-71139-5 Section 6.0
17.	PASS-1000, "Monitor Control Panel..." Screen Select: "RUN" button Select: "STOP" button Record Values in Appendix A Figure 10		
18.			EXP Laptop, Verify Voltage of HRF Rack
19.			Deactivate HRF Rack, EXPRESS Laptop per LS-71139-5 Section 6.0
20.	PEPSE Rack, Front panel Channel X Main: pwr sw - on Select: "DISARM" button NOTE: This step can be used on all ISPR channels (X= 1, 2, or 3).		
21.		ISPR Panel X Verify the following: • Power Indicator - off NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
22.	PEPSE Rack, upper front panel Select: "VOLTS" button Set Value: 116 V dc Verify the following: • "Volts" setting agrees with the value entered above.		
23.	PEPSE Rack, Front panel Channel X Main: Select: "ARM" button pwr sw - on NOTE: This step can be used on all ISPR channels (X = 1, 2, or 3).		
24.		ISPR Panel X Verify the following: • Power Indicator is orange NOTE: This step can be used on all ISPR channels (X= 1, 2, or 3).	

TABLE 6.15 POWER (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
25.	PEPSE Rack, Front Panel Channel X Main Record the following: • Voltage: _____ V dc NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
26.		ISPR Panel X, Using DVM Record the following: • TP1 Voltage: _____ V dc NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
27.			Activate HRF Rack, EXPRESS Laptop per LS-71139-5 Section 6.0
28.	PASS-1000 "Monitor Control Panel..." Screen Select: "RUN" button Select: "STOP" button Record Values in Appendix A Figure 10		
29.			EXP Laptop Verify Voltage of HRF Rack
30.			Deactivate HRF Rack, EXPRESS Laptop per LS-71139-5 Section 6.0
31.	PEPSE Rack, Front panel Channel X Main: pwr sw - on Select: "DISARM" button NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
32.		ISPR Panel X Verify the following: • Power Indicator - off NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
33.	PEPSE Rack, upper front panel Select: "VOLTS" button Set Value: 126 V dc Select: "ACCEPT" button Verify the following: • "Volts" setting agrees with the value entered above		
34.	PEPSE Rack, Front panel Channel X Main: Select: "ARM" button pwr sw - on NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
35.		ISPR Panel X Verify the following: • Power Indicator is orange NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
36.	PEPSE Rack, Front Panel Channel X Main Record the following: • Voltage: _____ V dc NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		

TABLE 6.15 POWER (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
37.		ISPR Panel X, Using DVM Record the following: • TP1 Voltage: _____ V dc NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
38.			Activate HRF Rack, EXP Laptop per LS-71139-5 Section 6.0
39.	PASS-1000 "Monitor Control Panel..." Screen Select: "RUN" button Select: "STOP" button Record Values in Appendix A Figure 10		
40.			EXP Laptop, Verify Voltage of HRF Rack
41.			Deactivate HRF Rack, EXPRESS Laptop per LS-71139-5 Section 6.0
42.	PEPSE Rack, Front panel Channel X Main: pwr sw _____ - on Select: "DISARM" button NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).		
43.		ISPR Panel X Verify the following: • Power Indicator _____ - off NOTE: This step can be used on all ISPR channels (X=1, 2, or 3).	
44.		Deactivate PASS-1000 per document LS71139-2, Section 6.0	
45.		Deactivate PRCU Wrkstn per document LS71139-2, Section 6.0	
46.	Deactivate Moderate Temperature Chiller per document LS71139-2, Section 6.0		
47.	Deactivate DAPC Rack per document LS-71139-2, Section 6.0		
48.	Deactivate Computational Rack per document LS-71139-2, Section 6.0		
49.	Deactivate PEPSE Rack per document LS-71139-2, Section 6.0		

6.2.2 Thermal

TBD

6.2.3 Standard Interface Rack (SIR)/Breakout Box (BOB)

The purpose of the SIR/BOB Test is to test the power and data interfaces of HRF Rack One between the RIC and the Payload drawer locations (PDL). The ability to configure, activate and utilize each drawer location and the Front Panel (deployed payload) location will be demonstrated for the following:

- PDL power
- PDL to RIC
- RIC to PDL RS-422
- Ethernet communications
- PDL to RIC
- RIC to PDL Discretes
- PDL to RIC Analog

NOTE: Unless otherwise stated, any of the following steps may be repeated and/or performed non-sequentially to accomplish the desired test objectives.

TABLE 6.16 SIR/BOB

Step	PRCU	HRF Rack 1	Payload
1.			Configure the Test Support System (TSS) hardware per drawing SEG38115983 NOTE: All IP addresses of 10.12.12 may be changed to 120.80.10 to be consistent with the HRF Rack RIC/IP addresses.
2.	Perform PRCU startup per LS-71139-2 Section 6.0		
3.		Perform HRF Rack Activation per LS-71139-5 Section 6.0	
4.		EXP Laptop; "EXPRESS – HRF" Screen Select: "EXPRESS Comm" button Select: "P/L Control" tab Select: "Payload Configuration" button Configure the Deployed Payload Location for Port Number 1900 (hex), Function Code 100, and Internet Protocol (IP) Address 10.12.12.100	
5.			TSS Laptop, left side panel Main pwr sw - on Verify the LED is illuminated
6.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "Settings" Select: "Control Panels" Verify the following: • "Control Panel" window appears
7.			TSS Laptop, "Control Panel" Window Select: "Network" Verify the following: • "Network" window appears
8.			TSS Laptop, "Network" Window Select: "TCP/IP -> 3 COM..." Verify the following: • "TCP/IP Properties" window appears

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
9.			<p>TSS Laptop, "TCP/IP Properties" Window IP Address: Set Value: "10.12.12.100" Select: "OK" button</p> <p>NOTE: If the IP Address is changed, the Laptop will need to be shutdown and power cycled for the change to take effect (Shutdown and power-off the laptop and then repeat the previous step and this step).</p>
10.		<p>EXP Laptop, "EXPRESS Comm: P/L Control" Screen Configure HRF Rack Locations 1 through 15 for Medium Rate Data Link (MRDL), RS-422 with Baud Rate 2400 and Function Codes 101 through 115, respectively</p>	
11.		<p>EXP Laptop, "EXPRESS Comm: P/L Control" Screen Configure Discretes as inputs to Solid State Power Control Module (SSPCM) (outputs from payload drawers) for HRF Rack Locations 1 through 15</p>	
12.		<p>EXP Laptop, "EXPRESS - Comm" Screen Select: "P/L Comm" tab Select: "Drawer Comm" button For all configured Locations Select: "Enable" radio button Select: "Execute" button</p> <p>NOTE: The following steps 16 through 79, will be repeated eight (8) times using the values in Table 6.17 for each iteration. The values from Table 6.17 are to be substituted for the corresponding bracketed '[]' labels for the appropriate iterations. The QA buys for the steps for each iteration as indicated and will be signed off in Table 6.18. This matrix will serve as the official record for completion of the steps.</p>	
13.		<p>HRF Rack, Front Panel Subrack Location [A]: Install BOB-1</p> <p>NOTE: Once drawer is installed, ensure lock handles are in place.</p>	
14.		<p>HRF Rack, Front Panel Subrack Location [B]: Install BOB-2</p> <p>NOTE: Once drawer is installed, ensure lock handles are in place.</p>	
15.		<p>EXP Laptop Verify continuity for the following:</p> <ul style="list-style-type: none"> • Subrack Location [A] • Subrack Location [B] 	
16.		<p>EXP Laptop, "EXPRESS EPS: Drawer Power" Screen For Locations [A] and [B] Select: "On" radio button Select: "Execute" button</p>	

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
17.		HRF Rack, Upper Front Panel Subrack [A] pwr sw - on Subrack [B] pwr sw - on Verify the following: <ul style="list-style-type: none"> Subrack [A] LED is illuminated green Subrack [B] LED is illuminated green 	
18.			TSS BOB 1, Front Panel Data pwr sw - on Verify the Data LED is illuminated red
19.			TSS BOB 2, Front Panel Data pwr sw - on Verify the Data LED is illuminated red
20.			TSS Signal Adapter Box (SAB), Front Panel Verify the LED is illuminated green
21.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "I&ES Simulator (Start).exe" Verify the following is displayed: <ul style="list-style-type: none"> "I&ES Configuration" Window
22.			TSS Laptop, "I&ES Configuration" Window Select: "Two Simulators" <u>BOB-1/Sim-1:</u> Select: Comm Protocol "TCP/IP" Set Value: TCP/IP Port "6400:" Set Value: Function Code "100" <u>BOB-2/Sim-2:</u> Select: Comm Protocol "RS-422" Set Value: Baud rate "2400" Set Value: Function Code "[C]" Select: "START" button Verify the following screens appear: <ul style="list-style-type: none"> "I&ES Sim-1 TCP/IP" "I&ES Sim-2 RS-422"
23.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Verify the following: <ul style="list-style-type: none"> HRF Rack Voltage $28 \pm 1V$ "Connection Established" is indicated "Transmission Control" LED is green
24.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: Select Request "10 " (data bytes) Set Value: Repetitions "2" Set Value: Frequency "1" Verify the following request in "I&ES to RIC Request (Outgoing)" field: <ul style="list-style-type: none"> 55AA 0018 009C 0100 0001 009C 0000 0000 0000 0001 0002

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
25.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen "I&ES to RIC Request (Outgoing)" field: Change the fifth word to value [D] Verify the following request is indicated: <ul style="list-style-type: none"> 55AA 0018 009C 0100 [D] 009C 0000 0000 0000 0001 0002
26.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Select: "SEND" button Verify the following: <ul style="list-style-type: none"> Two (2) messages sent above appear in the "RIC to I&ES Messages (Incoming)" field of Sim-2 (RS-422) <p>NOTE: An extra RS-422 checksum word is displayed at the end of the original message.</p>
27.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Discrete 1 & 2 Assignment "Write/Write" Select: "OK" button
28.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Discrete 1 & 2 Assignment "Write/Write" Select: "OK" button
29.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: output discretes (2 each) "High"
30.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Set Value: output discretes (2 each) "High"
31.	Health & Status Check Verify the following: <ul style="list-style-type: none"> SSPCM Discrete Status Word [E], check that bits [F] = 11 binary 		
32.	Health & Status Check Verify the following: <ul style="list-style-type: none"> SSPCM Discrete Status Word [G], check that bits [H] = 11 binary 		
33.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: output discretes (2 each) "Low" (0)
34.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Set Value: output discretes (2 each) "Low" (0)
35.	Health & Status Check Verify the following: <ul style="list-style-type: none"> SSPCM Discrete Status Word [E], check that bits [F] = 00 binary 		
36.	Health & Status Check SSPCM Discrete Status Word [G], check that bits [H] = 00 binary		

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
37.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Discrete 1 & 2 Assignment "Read/Read" Select: "OK" button
38.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Discrete 1 & 2 Assignment "Read/Read" Select: "OK" button
39.		EXP Laptop, "EXPRESS-EPS: Drawer Power" Screen For Locations [A] and [B] Select: "Off" radio button Select: "Execute" button	
40.		EXP Laptop, "EXPRESS-Comm: Drawer Comm" Screen For Locations [A] and [B] Select: "Inhibit" radio button Select: "Execute" button	
41.	PRCU, "Payload Command Table" Configure discretes as outputs from SSPCM (inputs to payload drawers) for HRF Rack Locations [A] and [B]		
42.		EXP Laptop, "EXPRESS-Comm: Drawer Comm" Screen For Locations [A] and [B] Select: "On" radio button Select: "Execute" button	
43.		EXP Laptop, "EXPRESS-Comm: Drawer Comm" Screen For Locations [A] and [B] Select: "Enable" radio button Select: "Execute" button	
44.	PRCU, "Payload Command Table" Set both of the HRF Rack SSPCM discrete outputs for HRF Rack Location [A] to "High" (1)		
45.	PRCU, "Payload Command Table" Set both of the HRF Rack SSPCM discrete outputs for HRF Rack Location [B] to "High" (1)		
46.	Health & Status Check SSPCM Discrete Status Word [E], check that bits [F] = 11 binary		
47.	Health & Status Check SSPCM Discrete Status Word [G], check that bits [H] = 11 binary		
48.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Verify discretes for Sim-1 are both indicating "1"
49.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Verify discretes for Sim-2 are both indicating "1"
50.		EXP Laptop, "EXPRESS Comm: P/L Control" Screen Set both of the HRF Rack SSPCM discrete outputs for HRF Rack Location [A] to "Low" (0)	

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
51.		EXP Laptop, "EXPRESS Comm: P/L Control" Screen Set both of the HRF Rack SSPCM discrete outputs for HRF Rack Location [B] to "Low" (0)	
52.	Health & Status Check SSPCM Discrete Status Word [E], check that bits [F] = 00 binary		
53.	Health & Status Check SSPCM Discrete Status Word [G], check that bits [H] = 00 binary		
54.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Verify discretes for Sim-1 are both indicating "0"
55.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Verify discretes for Sim-2 are both indicating "0"
56.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: Analog Output "-5V, Square wave @ 100% (-5V dc)"
57.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Set Value: Analog Output "-5V, Square wave @ 100% (-5V dc)"
58.	Health & Status Check Read and Record SSPCM Input Status Word [I]		
59.	Health & Status Check Read and Record SSPCM Input Status Word [J]		
60.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: Analog Output "+5V, Square wave @ 100% (+5V dc)"
61.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Set Value: Analog Output "+5V, Square wave @ 100% (+5V dc)"
62.	Health & Status Check Read and Record SSPCM Input Status Word [I]		
63.	Health & Status Check Read and Record SSPCM Input Status Word [J]		
64.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Select: "QUIT" button
65.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Select: "QUIT" button NOTE: Swap Ethernet/RS232 cables to appropriate BOB ports.
66.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "I&ES Simulator (Start).exe" Verify the following is displayed: <ul style="list-style-type: none"> "I&ES Configuration" Window Software on the TSS Laptop and configure for Two (2) Simulators as follows

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
67.			<p>TSS Laptop, "I&ES Configuration" Window</p> <p>Select: "Two Simulators"</p> <p><u>BOB-1/Sim-1:</u></p> <p>Select: Comm Protocol "RS-422"</p> <p>Set Value: Baud rate "2400"</p> <p>Set Value: Function Code "[K]"</p> <p><u>BOB-2/Sim-2:</u></p> <p>Select: Comm Protocol "TCP/IP"</p> <p>Set Value: TCP/IP Port "6400"</p> <p>Set Value: Function Code "100"</p> <p>Select: "START" button</p> <p>Verify the following screens appear:</p> <ul style="list-style-type: none"> • "I&ES Sim-1 RS-422" • "I&ES Sim-2 TCP/IP"
68.			<p>TSS Laptop, "I&ES Sim-1 RS-422" Screen</p> <p>Verify the following:</p> <ul style="list-style-type: none"> • HRF Rack Voltage $28 \pm 1V$
69.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Verify the following:</p> <ul style="list-style-type: none"> • "Connection Established" is indicated • "Transmission Control" LED is green
70.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Set Value: Select Request "10" (data bytes)</p> <p>Set Value: Repetitions "2"</p> <p>Set Value: Frequency "1"</p> <p>Verify the following request in "I&ES to RIC Request (Outgoing)" field:</p> <ul style="list-style-type: none"> • 55AA 0018 009C 0100 0001 009C 0000 0000 0000 0001 0002
71.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>"I&ES to RIC Request (Outgoing)" field: Change the fifth word to value [L]</p> <p>Verify the following request is indicated:</p> <ul style="list-style-type: none"> • 55AA 0018 009C 0100 [L] 009C 0000 0000 0000 0001 0002
72.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Select: "SEND" button</p> <p>Verify the following:</p> <ul style="list-style-type: none"> • Two (2) messages sent above appear in the "RIC to I&ES Messages (Incoming)" field of Sim-1 (RS-422) <p>NOTE: An extra RS-422 checksum word is displayed at the end of the original message.</p>
73.			<p>TSS Laptop, "I&ES Sim-1 RS-422" Screen</p> <p>Select: "QUIT" button</p>

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
74.			TSS Laptop, "I&ES Sim-2 TCP/IP" Screen Select: "QUIT" button NOTE: Swap Ethernet/RS232 cables to appropriate BOB ports.
75.			TSS BOB 1, Front Panel Data pwr sw - off Verify the Data LED is not illuminated
76.			TSS BOB 2, Front Panel Data pwr sw - off Verify the Data LED is not illuminated
77.		HRF Rack, Upper Front Panel Subrack [A] pwr sw - off Subrack [B] pwr sw - off Verify the following: <ul style="list-style-type: none"> Subrack [A] LED is not illuminated Subrack [B] LED is not illuminated 	
78.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen For Locations [A] and [B] Select: "Off" radio button Select: "Execute" button NOTE: The QA buys for the following steps indicated in Table 6.18 and signed off once performed. This matrix will serve as the official record for completion of the steps.	
79.		EXP Laptop, "EXPRESS Comm: Drawer Comm" Screen For all configured Locations Select: "Inhibit" radio button Select: "Execute" button	
80.		EXP Laptop, Front Panel Configure Discretes as inputs for SSPCM (outputs from the payload) for the Deployed Location NOTE: The following six (6) steps must be performed in proper order and as a unit to support the Payload A2 Location testing.	
81.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen Verify Payload A2 Location power deactivated.	
82.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "Shutdown" Verify the following: <ul style="list-style-type: none"> "Shutdown" window appears
83.			TSS Laptop, "Shutdown" Window Select: "YES" button Verify the following: <ul style="list-style-type: none"> Laptop is powered off
84.		Configure HRF Rack Location J1 for MRDL, Ethernet with Port Number 1900 (hex), Function Code 108 and IP Address 10.12.12.108	

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
85.		HRF Rack, Front panel Subrack Location J1: Verify BOB-1 installed Subrack Location J2: Verify BOB-2 installed	
86.		HRF Rack, Upper front panel Subrack J1 pwr sw - off Subrack J2 pwr sw - off Verify the following: <ul style="list-style-type: none"> Subrack [A] LED is not illuminated Subrack [B] LED is not illuminated 	
87.			Disconnect the following hardware: <ul style="list-style-type: none"> SAB/BOB Data Interface Cable from BOB-1 SAB/BOB Data Interface Cable from SAB (BOB 1 Connector)
88.			Connect the following hardware: <ul style="list-style-type: none"> SAB/Front Panel Payload Interface Cable SEG38116524-301 to SAB (BOB-1 Connector) SAB/Front Panel Payload Interface Cable SEG38116524-301 to HRF Rack Front Panel Payload Data (RCP J2) and Power (RCP J1)
89.		EXP Laptop, "EXPRESS Comm: Drawer Comm" Screen Locations Payload A2, J1 and J2 Select: "Enable" radio button Select: "Execute" button	
90.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen Location J2 Select: "On" radio button Select: "Execute" button	
91.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen Payload A2 Location Select: "On" radio button Select: "Execute" button	
92.		HRF Rack, Upper Front Panel Payload pwr sw - on Verify the following: <ul style="list-style-type: none"> Payload LED is illuminated green 	
93.			TSS SAB, Front Panel Verify the following: LED is illuminated green
94.		HRF Rack, Upper Front Panel Subrack J2 pwr sw - on Verify the following: <ul style="list-style-type: none"> Subrack J2 LED is illuminated green 	
95.			TSS BOB-2, Front Panel Data sw - on Verify the following: Data LED is illuminated red
96.			TSS Laptop, Left side panel Main pwr sw - on
97.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "I&ES Simulator (Start).exe" Verify the following is displayed: <ul style="list-style-type: none"> "I&ES Configuration" Window

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
98.			<p>TSS Laptop, "I&ES Configuration" Window</p> <p>Select: "Two Simulators"</p> <p><u>BOB-1/Sim-1:</u></p> <p>Select: Comm Protocol "TCP/IP"</p> <p>Set Value: TCP/IP Port "6400"</p> <p>Set Value: Function Code "100"</p> <p><u>BOB-2/Sim-2:</u></p> <p>Select: Comm Protocol "RS-422"</p> <p>Set Value: Baud rate "2400"</p> <p>Set Value: Function Code "115"</p> <p>Select: "START" button</p> <p>Verify the following screens appear:</p> <ul style="list-style-type: none"> • "I&ES Sim-1 TCP/IP" • "I&ES Sim-2 RS-422"
99.			<p>TSS Laptop, "I&ES Sim-1 TCP/IP" Screen</p> <p>Verify the following:</p> <ul style="list-style-type: none"> • HRF Rack Voltage $28 \pm 1V$ • "Connection Established" is indicated • "Transmission Control" LED is green
100.			<p>TSS Laptop, "I&ES Sim-1 TCP/IP" Screen</p> <p>Set Value: Select Request "10" (data bytes)</p> <p>Set Value: Repetitions "2"</p> <p>Set Value: Frequency "1"</p> <p>Verify the following request in "I&ES to RIC Request (Outgoing)" field:</p> <ul style="list-style-type: none"> • 55AA 0018 009C 0100 0001 009C 0000 0000 0000 0001 0002
101.			<p>TSS Laptop, "I&ES Sim-1 TCP/IP" Screen</p> <p>"I&ES to RIC Request (Outgoing)" field:</p> <p>Change the fifth word to value '0115'</p> <p>Verify the following request is indicated:</p> <ul style="list-style-type: none"> • 55AA 0018 009C 0100 0115 009C 0000 0000 0000 0001 0002
102.			<p>TSS Laptop, "I&ES Sim-1 TCP/IP" Screen</p> <p>Select: "SEND" button</p> <p>Verify the following:</p> <ul style="list-style-type: none"> • Two (2) messages sent above appear in the "RIC to I&ES Messages (Incoming)" field of Sim-2 (RS-422) <p>NOTE: An extra RS-422 checksum word is displayed at the end of the original message.</p>
103.			<p>TSS Laptop, "I&ES Sim-1 TCP/IP" Screen</p> <p>Discrete 1 & 2 Assignment "Write/Write"</p> <p>Select: "OK" button</p>

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
104.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: output discretes (2 each) "High" (1)
105.	Health & Status Check Verify the following: <ul style="list-style-type: none"> SSPCM Discrete Status Word 19, check that bits 2, 1 = 11 binary 		
106.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: output discretes (2 each) "Low" (0)
107.	Health & Status Check Verify the following: <ul style="list-style-type: none"> SSPCM Discrete Status Word 19, check that bits 2, 1 = 00 binary 		
108.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Discrete 1 & 2 Assignment "Read/Read" Select: "OK" button
109.		EXP Laptop, "EXPRESS-EPS: Drawer Power" Screen For Payload A2 Location Select: "Off" radio button Select: "Execute" button	
110.		EXP Laptop, "EXPRESS-Comm: Drawer Comm" Screen For Payload A2 Location Select: "Inhibit" radio button Select: "Execute" button	
111.		EXP Laptop, Front Panel Configure Discretes as outputs from SSPCM (inputs to the payload) for the Payload A2 Payload Location	
112.		EXP Laptop, "EXPRESS-EPS: Drawer Power" Screen For Payload A2 Location Select: "On" radio button Select: "Execute" button	
113.		EXP Laptop, "EXPRESS-Comm: Drawer Comm" Screen For Payload A2 Location Select: "Enable" radio button Select: "Execute" button	
114.		EXP Laptop, Front Panel Set both of the HRF Rack SSPCM discrete outputs for the Payload A2 Location to "High" (1)	
115.	Health & Status Check SSPCM Discrete Status Word 19, check that bits 2, 1 = 11 binary		
116.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Verify the following: <ul style="list-style-type: none"> Discretes for Simulator 1 are both indicating "1"
117.		EXP Laptop, Front Panel Set both of the HRF Rack SSPCM discrete outputs for the Payload A2 Location to "Low" (0)	
118.	Health & Status Check SSPCM Discrete Status Word 19, check that bits 2, 1 = 00 binary		

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
119.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Discretes for Simulator 1 are both indicating "0"
120.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: Analog Output "-5V, Square wave @ 100% (-5V dc)"
121.	Health & Status Check Record the value of the SSPCM Input Status Word 40 in the Table 6.18		
122.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Set Value: Analog Output "+5V, Square wave @ 100% (+5V dc)"
123.	Health & Status Check Record the value of the SSPCM Input Status Word 40 in the Table 6.18		
124.			TSS Laptop, "I&ES Sim-1 TCP/IP" Screen Select: "QUIT" button
125.			TSS Laptop, "I&ES Sim-2 RS-422" Screen Select: "QUIT" button
126.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "Settings" Select: "Control Panels" Verify the following: • "Control Panel" window appears
127.			TSS Laptop, "Control Panel" Window Select: "Network" Verify the following: • "Network" window appears
128.			TSS Laptop, "Network" Window Select: "TCP/IP -> 3 COM..." Verify the following: • "TCP/IP Properties" window appears
129.			TSS Laptop, "TCP/IP Properties" Window IP Address: Set Value: "10.12.12.108" Select: "OK" button
130.			TSS Laptop, "Network" Window Select: "OK" button
131.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "Shutdown" Verify the following: • "Shutdown" window appears
132.			TSS Laptop, "Shutdown" Window Select: "YES" button Verify the following: • TSS Laptop powers off
133.		HRF Rack, Upper Front Panel Payload pwr sw - off Verify the following: • Payload LED is not illuminated	
134.			TSS SAB, Front Panel Verify the following: LED is not illuminated

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
135.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen Payload A2 Location Select: "Off" radio button Select: "Execute" button	
136.		EXP Laptop, "EXPRESS Comm: Drawer Comm" Screen Payload A2 Location Select: "Inhibit" radio button Select: "Execute" button	
137.		EXP Laptop, "EXPRESS Comm: P/L Control" Screen Configure the Payload A2 Location for MRDL, RS-422 with Baud Rate 2400 and Function Code 100	
138.		EXP Laptop, "EXPRESS EPS: Drawer Power" Screen Payload A2 Location Select: "On" radio button Select: "Execute" button	
139.		EXP Laptop, "EXPRESS Comm: Drawer Comm" Screen Payload A2 Location Select: "Enable" radio button Select: "Execute" button	
140.		HRF Rack, Upper Front Panel Payload pwr sw - on Verify the following: • Payload LED is illuminated green	
141.			TSS SAB, Front Panel Verify the following: • Power LED is illuminated green
142.			TSS Laptop, Left side panel Main pwr sw - on
143.			TSS Laptop, "Windows 95" desktop Select: "Start" menu Select: "I&ES Simulator (Start).exe" Verify the following is displayed: • "I&ES Configuration" Window
144.			TSS Laptop, "I&ES Configuration" Window Select: "Two Simulators" <u>BOB-1/Sim-1:</u> Select: Comm Protocol "RS-422" Set Value: Baud rate "2400" Set Value: Function Code "100" <u>BOB-2/Sim-2:</u> Select: Comm Protocol "TCP/IP" Set Value: TCP/IP Port "6400" Set Value: Function Code "108" Select: "START" button Verify the following Screens appear: • "I&ES Sim-1 RS-422" • "I&ES Sim-2 TCP/IP"
145.			TSS Laptop, "I&ES Sim-1 RS-422" Screen Verify the following: • HRF Rack Voltage $28 \pm 1V$
146.			TSS Laptop, "I&ES Sim-2 TCP/IP" Screen Verify the following: • "Connection Established" is indicated • "Transmission Control" LED is green

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
147.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Set Value: Select Request "10" (data bytes)</p> <p>Set Value: Repetitions "2"</p> <p>Set Value: Frequency "1"</p> <p>Verify the following request in "I&ES to RIC Request (Outgoing)" field:</p> <ul style="list-style-type: none"> 55AA 0018 009C 0108 0001 009C 0000 0000 0000 0001 0002
148.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>"I&ES to RIC Request (Outgoing)" field: Change the fifth word to value '0100'</p> <p>Verify the following request is indicated:</p> <ul style="list-style-type: none"> 55AA 0018 009C 0108 0100 009C 0000 0000 0000 0001 0002
149.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Select: "SEND" button</p> <p>Verify the following:</p> <ul style="list-style-type: none"> Two (2) messages sent above appear in the "RIC to I&ES Messages (Incoming)" field of Sim-1 (RS-422) <p>NOTE: An extra RS-422 checksum word is displayed at the end of the original message.</p>
150.			<p>TSS Laptop, "I&ES Sim-1 RS-422" Screen</p> <p>Select: "QUIT" button</p>
151.			<p>TSS Laptop, "I&ES Sim-2 TCP/IP" Screen</p> <p>Select: "QUIT" button</p>
152.			<p>TSS BOB-2, front panel</p> <p>Data sw - off</p> <p>Verify the following:</p> <ul style="list-style-type: none"> Data LED is not illuminated
153.		<p>HRF Rack, Upper Front Panel</p> <p>Payload pwr sw - off</p> <p>Subrack J2 pwr sw - off</p> <p>Verify the following:</p> <ul style="list-style-type: none"> Subrack J1 LED is not illuminated Subrack J2 LED is not illuminated <p>Verify Pwr SAB LED - off</p>	
154.			<p>TSS SAB, Front panel</p> <p>Verify the following:</p> <ul style="list-style-type: none"> Pwr LED is not illuminated
155.		<p>EXP Laptop, "EXPRESS EPS: Drawer Power" Screen</p> <p>For Location J2, and Payload A2</p> <p>Select: "Off" radio button</p> <p>Select: "Execute" button</p>	

TABLE 6.16 SIR/BOB (CONT'D)

Step	PRCU	HRF Rack 1	Payload
157.			TSS Laptop, "Windows 95" desktop Select: "START" menu Select: "Shutdown" Verify the following: • "Shutdown" window appears
158.			TSS Laptop, "Shutdown" Window Select: "YES" button
159.		HRF Rack, Front Panel Subrack Location J1: Remove BOB-1 NOTE: When sliding drawer out of Subrack location, be sure to lift safety stop on the left side of the payload drawer.	
160.		HRF Rack, Front Panel Subrack Location J2: Remove BOB-2 NOTE: When sliding drawer out of Subrack location, be sure to lift safety stop on the left side of the payload drawer.	
161.			Disconnect the following hardware: • SAB/Front Panel Payload Interface Cable SEG38116524-301 to SAB (BOB-1) • SAB/Front Panel Payload Interface Cable SEG38116524-301 to HRF Rack Front Panel Payload Data (RCP J2) and Power (RCP J1)
162.			Disconnect the following hardware: • TSS Assy
163.		Perform HRF Rack Deactivation per LS-71139-5 Section 6.0	
164.	Perform PRCU shutdown per LS-71139-2 Section 6.0		

TABLE 6.17 ITERATION GUIDE FOR SIR/BOB VALUES

Iteration	[A]	[B]	[C]	[D]	[E]	[F]	[G]	[H]	[I]	[J]	[K]	[L]
1.	B1	C2	109	0109	16	(1, 0)	17	(9, 8)	25	41	101	0101
2.	C1	C2	109	0109	16	(4, 3)	17	(9, 8)	26	41	102	0102
3.	D1	D2	110	0110	16	(7, 6)	18, 17	(0, 11)	29	42	103	0103
4.	E1	E2	111	0111	16	(10, 9)	18	(3, 2)	31	43	104	0104
5.	F1	F2	112	0112	16	(13, 12)	18	(6, 5)	33	44	105	0105
6.	G1	G2	113	0113	17, 16	(0, 15)	18	(9, 8)	35	45	106	0106
7.	H1	H2	114	0114	17	(3, 2)	18	(12, 11)	37	46	107	0107
8.	J1	J2	115	0115	17	(6, 5)	18	(15, 14)	39	47	108	0108

Key for Table Elements:

BOB-1 is associated with [A] [E] [F] [I] [K] [L]

BOB-2 is associated with [B] [C] [D] [G] [H] [J]

[A] [B] – Rack Drawer Locations

[C] [K] – Function Code Assignments

[D] [L] – Destination Function Codes

[E] [F] [G] [H] – H&S Discrete Status Words and bits per Drawer Location

[I] [J] – H&S (Analog) Input Status Words

TABLE 6.18 QUALITY BUY-OFF MATRIX FOR MIPS

Iteration 1	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	T																	
	QA																	
Iteration 2	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	T																	
	QA																	
Iteration 3	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	T																	
	QA																	
Iteration 4	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	T																	
	QA																	
Iteration 5	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	T																	
	QA																	

TABLE 6.18 QUALITY BUY-OFF MATRIX FOR MIPS (CONT'D)

Iteration 6	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	Tech.																	
	Quality																	
Iteration 7	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	Tech.																	
	Quality																	
Iteration 8	Steps:	15.	18.	19.	26.	31.	32.	35.	36.	48.	49.	54.	55.	58.	59.	62.	63.	72.
	Tech.																	
	Quality																	
Deployed	Steps:	86.	88.	93.	102.	105.	107.	116.	119.	121.	123.	149.						
	Tech.																	
	Quality																	

6.3 OFFLINE PAYLOAD FUNCTIONALS

6.3.1 Gas Analyzer System for Metabolic Analysis of Physiology (GASMAP)

6.3.1.1 GASMAP Analyzer Module Functional

This sequence describes the functional procedure performed on the HRF GASMAP units. This procedure is performed with the following assumptions:

- The analyzer has been calibrated
- The functionality of the GASMAP payload including the following subsystems:
 - Analyzer Roughing
 - System Power
 - Front Panel Controls
 - Identification
 - Environmental Sensor
 - Flow Meter
 - Room Air Measurement
 - Gas Delivery System

The following list provides the flight units or equivalent hardware that may be used to perform off line functional tests. The part numbers and serial numbers will be noted on the TPS.

TABLE 6.19 GASMAP ANALYZER MODULE HARDWARE

Qty	Item	Class
1	Analyzer Module	Flight
1	Catheter	Flight
1	Flowmeter Cable	Flight
1	Flow Cartridge	Flight
1	Aux Power Cable	GSE

Once the GASMAP payload has successfully completed functional testing, it can be integrated into the HRF Rack.

TABLE 6.20 GASMAP FUNCTIONAL CHECKOUT

Step	Payload	Tech																
1.	<p>Remove the following hardware from Building 241 Controlled Storage:</p> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>GASMAP Analyzer Module</td></tr><tr><td>1</td><td>Catheter</td></tr><tr><td>1</td><td>Flowmeter Cable</td></tr><tr><td>1</td><td>Flow Cartridge</td></tr><tr><td>1</td><td>AUX Power Cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></tbody></table>	Qty	Item	1	GASMAP Analyzer Module	1	Catheter	1	Flowmeter Cable	1	Flow Cartridge	1	AUX Power Cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item																	
1	GASMAP Analyzer Module																	
1	Catheter																	
1	Flowmeter Cable																	
1	Flow Cartridge																	
1	AUX Power Cable																	
1	Variable Power Supply																	
1	Digital Multimeter																	
2.	Assemble the hardware from above Figure 1 in Appendix B																	
3.	<p>GASMAP Analyzer Module, Front Panel</p> <p>Verify the following:</p> <ul style="list-style-type: none">• “HRF GASMAP MAIN/AUX” sw - AUX• “Main Power” sw - off• “AUX Power” sw - off																	
4.		<p>Variable Power Supply</p> <p>Verify the following:</p> <p>Main pwr sw - off</p> <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p>																
5.		<p>Connect the following hardware:</p> <ul style="list-style-type: none">• AUX Power Cable (GSE) to Variable Power Supply• Variable Power Supply to Power Source																
6.		<p>Variable Power Supply, front panel</p> <p>Main pwr sw - on</p> <p>Set Value: 28 V</p> <p>Set Value: 25 amps</p> <p>NOTE: If using a calibrated power supply, continue on to the following step.</p> <p>Using Digital Multimeter:</p> <ul style="list-style-type: none">• Verify the voltage between pins 1 and 2 measures 28 ± 0.5 V dc <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p>																
7.		<p>Variable Power Supply</p> <p>Main pwr sw - off</p>																
8.	<p>Connect the following hardware:</p> <ul style="list-style-type: none">• AUX Power Cable (GSE) to GASMAP Analyzer Module																	
9.		<p>Variable Power Supply, front panel</p> <p>Main Power sw - on</p> <p>Set Value: 28 V</p> <p>Set Value: 25 amps</p>																

TABLE 6.20 GASMAP FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
10.	<p>GASMAP Analyzer Module, Front Panel</p> <p>"HRF GASMAP MAIN/AUX" sw - AUX</p> <p>"HRF GASMAP" sw - on</p> <p>NOTE: The front panel display may take up to one (1) minute to activate. The GASMAP goes through a pressure check to determine if the analyzer has maintained vacuum. This process can take anywhere from ten (10) seconds to thirty (30) minutes depending upon the state of the payload. When this process is complete, continue with the rest of the sequence.</p> <p>Verify the following appears:</p> <ul style="list-style-type: none"> • "Main Menu" Screen • Error LED is not illuminated • "POWER" LED is illuminated • HRF GASMAP cooling fan is operational • Air is blowing from the rear panel of the payload <p>NOTE: If the Error LED is illuminated, perform sequence 6.3.1.3 in its entirety.</p>	
11.	<p>GASMAP Analyzer Module, Front Panel</p> <p>Verify the following appears:</p> <ul style="list-style-type: none"> • "Main Menu" Screen • Error LED is not illuminated 	
12.	<p>GASMAP Analyzer Module, "Main Menu" Screen</p> <p>Select: "Diag" <3></p> <p>Select: "Display" <1></p> <p>Select: "Devices" <3></p> <p>Select: "Ion Pump" <1></p> <p>Verify the following appears:</p> <ul style="list-style-type: none"> • "Ion Pump" Screen 	
13.	<p>GASMAP Analyzer Module, "Ion Pump" Screen</p> <p>Record the following measurements:</p> <p>IPC = _____ μA ($< 20 \mu\text{A}$)</p> <p>IPV = _____ V (4500 – 6300 V)</p>	
14.	<p>GASMAP Analyzer Module, Front Panel</p> <p>Select: <MAIN MENU></p> <p>Verify the following appears:</p> <ul style="list-style-type: none"> • "Main Menu" Screen 	
15.	<p>GASMAP Analyzer Module, "Main Menu" Screen</p> <p>Select: "Diag" <3></p> <p>Select: "Test" <3></p> <p>Select: "LCD" <2></p> <p>Verify the following:</p> <ul style="list-style-type: none"> • All LCD segments are functional 	
16.	<p>GASMAP Analyzer Module, "LCD Test" Screen</p> <p>Select: <MAIN MENU></p> <p>Verify the following appears:</p> <ul style="list-style-type: none"> • "Main Menu" Screen 	

TABLE 6.20 GASMAP FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
17.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Test" <3> Select: "Lamp" <4> Verify the following appears: <ul style="list-style-type: none"> "Diagnostic Test" Screen 	
18.	GASMAP Analyzer Module, "Lamp Test" Screen Select: <1> Select: <2> Verify the following: <ul style="list-style-type: none"> "ERROR" LED is on "CHECK LCD" LED is on 	
19.	GASMAP Analyzer Module, "Lamp Test" Screen Select: <1> Select: <2> Verify the following: <ul style="list-style-type: none"> "ERROR" LED is off "CHECK LCD" LED is off 	
20.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Test" <3> Select: "Keypad" <1> Verify the following is displayed: <ul style="list-style-type: none"> "Key Pad Test" Screen 	
21.	GASMAP Analyzer Module, "Key Pad Test" Screen Select each key three (3) times, except <ESC> Verify the following: <ul style="list-style-type: none"> All keys are functional 	
22.	GASMAP Analyzer Module, "Key Pad Test" Screen Select: <ESC> Select: <MAIN MENU> Verify the following appears: <ul style="list-style-type: none"> "Main Menu" Screen 	
23.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Display" <1> Select: "System" <1> Select: "About" <1> Verify the following appears: <ul style="list-style-type: none"> "About HRF GASMAP" Screen 	
24.	GASMAP Analyzer Module, "About HRF" Screen Record the following information: HRF GASMAP S/N: _____ HRF GASMAP Information System (IS) S/W Version: _____ HRF GASMAP RAMS S/W Version: _____ HRF GASMAP Ethernet Address: _____ HRF GASMAP IP Address: _____ Select: <MAIN MENU> Verify the following appears: <ul style="list-style-type: none"> "Main Menu" Screen 	
25.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Display" <1> Select: "Devices" <3> Select: <↓> Select: "Environment" <2> Verify the following appears: <ul style="list-style-type: none"> "Environment" Screen 	

TABLE 6.20 GASMAP FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
26.	GASMAP Analyzer Module, "Environment" Screen Record the following information: <ul style="list-style-type: none"> Cabin Temperature: _____ °C Barometric Pressure: _____ mmHg 	
27.		Record the following information: <ul style="list-style-type: none"> Ambient Temperature: _____ °C Barometric Pressure: _____ mmHg
28.	GASMAP Analyzer Module, Cabin Temperature Sensor Place finger on sensor for 10-20 seconds Verify the following: <ul style="list-style-type: none"> Sensor temperature increases 	
29.	GASMAP Analyzer Module, Cabin Temperature Sensor Remove finger from sensor Verify the following: <ul style="list-style-type: none"> Sensor temperature decreases Select: <MAIN MENU> Verify the following appears: <ul style="list-style-type: none"> "Main Menu" Screen 	
30.	Connect the following hardware: <ul style="list-style-type: none"> Turbine flow meter (GSE) to GASMAP Analyzer Module, J6 	
31.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Test" <3> Select: "VMM" <3> Verify the following appears: <ul style="list-style-type: none"> "VMM Test" Screen 	
32.	GASMAP Analyzer Module, "VMM Test" Screen Breathe into the flow meter: <ul style="list-style-type: none"> Perform two (2) slow inhales Verify the following: Front panel inhale value increases during inhales	
33.	GASMAP Analyzer Module, "VMM Test" Screen Breathe into the flow meter: <ul style="list-style-type: none"> Perform two (2) slow exhales Verify the following: Front panel exhale value increases during exhales	
34.	GASMAP Analyzer Module, "VMM Test" Screen Select: <MAIN MENU> Verify the following is displayed: <ul style="list-style-type: none"> "Main Menu" Screen 	
35.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Diag" <3> Select: "Setup" <2> Select: "Sam Drw" <5> Verify the following appears: <ul style="list-style-type: none"> "Sample Draw" Screen 	
36.	GASMAP Analyzer Module, "Sample Draw" Screen Select: "Cath Pump" <1> Select: <ENTER> Verify the following appears: <ul style="list-style-type: none"> "Diag Setup" Screen 	
37.	GASMAP Analyzer Module, "Diag Setup" Screen Select: <MAIN MENU> Verify the following appears: <ul style="list-style-type: none"> "Main Menu" Screen 	
38.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Operate" <4> Select: "Default1" <1> Verify the following appears: <ul style="list-style-type: none"> "Atmospheric Information" Screen 	
NOTE: Wait thirty (30) seconds before proceeding.		

TABLE 6.20 GASMAP FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech																				
39.	<p>GASMAP Analyzer Module, "Atmospheric Information" Screen</p> <p>Record the following information:</p> <table><thead><tr><th>Gas</th><th>Room Air</th><th>Measured</th><th>Limits</th></tr></thead><tbody><tr><td>N₂</td><td>78.08%</td><td>_____</td><td>77.08<X<79.08%</td></tr><tr><td>O₂</td><td>20.93%</td><td>_____</td><td>19.93<X<21.93%</td></tr><tr><td>CO₂</td><td>0.05%</td><td>_____</td><td>0<X<0.25%</td></tr><tr><td>Ar</td><td>0.94%</td><td>_____</td><td>0.64<X<1.24%</td></tr></tbody></table>	Gas	Room Air	Measured	Limits	N ₂	78.08%	_____	77.08<X<79.08%	O ₂	20.93%	_____	19.93<X<21.93%	CO ₂	0.05%	_____	0<X<0.25%	Ar	0.94%	_____	0.64<X<1.24%	
Gas	Room Air	Measured	Limits																			
N ₂	78.08%	_____	77.08<X<79.08%																			
O ₂	20.93%	_____	19.93<X<21.93%																			
CO ₂	0.05%	_____	0<X<0.25%																			
Ar	0.94%	_____	0.64<X<1.24%																			
40.	<p>GASMAP Analyzer Module, "Atmospheric Information" Screen</p> <p>Select: <MAIN MENU></p> <p>Verify the following appears:</p> <ul style="list-style-type: none">• "Main Menu" Screen																					
41.	<p>GASMAP Analyzer Module, "Main Menu" Screen</p> <p>Select: "Diag" <3></p> <p>Select: "Display" <1></p> <p>Select: "Devices" <3></p> <p>Select: "CathPmp" <2></p> <p>Verify the following appears:</p> <ul style="list-style-type: none">• "Catheter Pump" Screen																					
42.	<p>GASMAP Analyzer Module, "Catheter Pump" Screen</p> <p>Record the following information:</p> <ul style="list-style-type: none">• Catheter Flow: _____ ml/min (60 ± 5 ml/min)• Place finger over end of catheter to block flow <p>Record the following information:</p> <ul style="list-style-type: none">• Catheter Flow: _____ ml/min (≤ 10 ml/min) <p>Verify the following:</p> <ul style="list-style-type: none">• "ERROR" LED is on																					
43.	<p>GASMAP Analyzer Module, "Catheter Pump" Screen</p> <p>Select: <MAIN MENU></p> <p>Verify the following appears:</p> <ul style="list-style-type: none">• "Main Menu" Screen																					
44.	<p>GASMAP Analyzer Module, "Main Menu" Screen</p> <p>Perform Section 6.3.1.3 to clear "ERROR" LED</p> <p>NOTE: Allow GASMAP to run for at least one (1) hour before powering down unit.</p> <p>Record the following values prior to deactivation:</p> <p>IPC: _____ μA</p> <p>IPV: _____ V</p>																					
45.	<p>GASMAP Analyzer Module, Front Panel</p> <p>"Main Power" sw - off</p> <p>"HRF GASMAP MAIN/AUX" sw - MAIN</p>																					
46.		<p>Variable Power Supply</p> <p>Main pwr sw - off</p>																				
47.		<p>Disconnect the following hardware:</p> <ul style="list-style-type: none">• Variable Power Supply from Power Source• AUX Power Cable (GSE) from GASMAP Analyzer Module (J8)• AUX Power Cable (GSE) from Variable Power Supply																				
48.	<p>Disassemble All GASMAP hardware</p> <p>Return the following hardware to Building 241 Controlled Storage:</p> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>GASMAP Analyzer Module</td></tr><tr><td>1</td><td>Catheter</td></tr><tr><td>1</td><td>Flowmeter Cable</td></tr><tr><td>1</td><td>Flow Cartridge</td></tr><tr><td>1</td><td>AUX Power Cable</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></tbody></table>	Qty	Item	1	GASMAP Analyzer Module	1	Catheter	1	Flowmeter Cable	1	Flow Cartridge	1	AUX Power Cable	1	Digital Multimeter							
Qty	Item																					
1	GASMAP Analyzer Module																					
1	Catheter																					
1	Flowmeter Cable																					
1	Flow Cartridge																					
1	AUX Power Cable																					
1	Digital Multimeter																					

6.3.1.2 GASMAP Calibration Module Functional

This section describes the procedures for the functional test of the GASMAP Calibration Module. The following table describes the flight units or equivalent hardware that may be used to perform off-line functional tests.

TABLE 6.21 GASMAP CALIBRATION MODULE HARDWARE

Qty	Item	Class
1	GASMAP Calibration Module	Flight

The objective of this test is to assess the functionality of the GASMAP Calibration Module after integration into the HRF Rack.

TABLE 6.22 GASMAP CALIBRATION FUNCTIONAL

Step	Payload	Tech																
1.	<p>Remove the following hardware from Building 241 Controlled Storage:</p> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>GASMAP Calibration Module</td></tr><tr><td>1</td><td>Catheter</td></tr><tr><td>1</td><td>GSE Calibration Solenoid Test Box</td></tr><tr><td>1</td><td>External Low Pressure Gauge</td></tr><tr><td>1</td><td>Power Cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></tbody></table>	Qty	Item	1	GASMAP Calibration Module	1	Catheter	1	GSE Calibration Solenoid Test Box	1	External Low Pressure Gauge	1	Power Cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item																	
1	GASMAP Calibration Module																	
1	Catheter																	
1	GSE Calibration Solenoid Test Box																	
1	External Low Pressure Gauge																	
1	Power Cable																	
1	Variable Power Supply																	
1	Digital Multimeter																	
2.	<p>GASMAP Calibration Module, Front Panel</p> <p>Verify the following:</p> <ul style="list-style-type: none">Valve 1 - closedValve 2 - closedValve 3 - closed																	
3.		<p>GSE Calibration Solenoid Test Box</p> <p>Verify the following:</p> <ul style="list-style-type: none">SW1 - offSW2 - offSW3 - offSW4 - off																
4.		<p>Variable Power Supply</p> <p>Verify the following:</p> <p>Main pwr sw - off</p> <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p>																
5		<p>Connect the following hardware:</p> <ul style="list-style-type: none">Power Cable (GSE) to Variable Power SupplyVariable Power Supply to Power Source																

TABLE 6.22 GASMAP CALIBRATION FUNCTIONAL (CONT'D)

Step	Payload	Tech
6.		Variable Power Supply, front panel Main Power sw - on Set Value: 12 V Set Value: 2 amps NOTE: If using a calibrated power supply, continue on with the following step. Using Digital Multimeter: <ul style="list-style-type: none"> Verify the voltage between pins 1 and 2 measures 12 ± 0.5 V dc Record the following information: Model: _____ NASA Tag #: _____ Calibration #: _____ Calibration Date: _____
7.		Variable Power Supply Main pwr sw - off
8.		Connect the following hardware: GSE Calibration Solenoid Test Box to Variable Power Supply
9.	GASMAP Calibration Module, Front Panel Connect the following: <ul style="list-style-type: none"> GSE Calibration Solenoid Test Box, P1 to GASMAP Calibration Module, J1 (SOL POWER) 	
10.		Variable Power Supply, front panel Main Power sw - on Set Value: 12 V Set Value: 2 amps
11.	GASMAP Calibration Module, Front Panel Connect the following: <ul style="list-style-type: none"> GASMAP Calibration Module (sample output) to Catheter 	
12.	GASMAP Calibration Module, Front Panel <ul style="list-style-type: none"> Valve 1 - open Valve 2 - open Valve 3 - open NOTE: Valves can be opened by rotating knobs in a direction clockwise one (1) to three (3) times.	
13.	GASMAP Calibration Module, Front Panel Verify/Record the following information: Tank 1 pressure: _____ psi ($400 \leq P \leq 1450$) Tank 2 pressure: _____ psi ($400 \leq P \leq 1450$) Tank 3 pressure: _____ psi ($400 \leq P \leq 1450$)	
14.		GSE Calibration Solenoid Test Box <ul style="list-style-type: none"> SW1 - on SW2 - off SW3 - off SW4 - off Verify the following: SOL1 LED is illuminated Audible click can be heard NOTE: Allow fifteen (15) seconds for the pressure to stabilize before continuing with the following step.
15.	External Low Pressure Gauge, Front Panel Record the following information: <ul style="list-style-type: none"> Tank 1 pressure: _____ psi ($6 \leq P \leq 9$) 	

TABLE 6.22 GASMAP CALIBRATION FUNCTIONAL (CONT'D)

Step	Payload	Tech
16.		GSE Calibration Solenoid Test Box • SW1 - off • SW2 - off • SW3 - off • SW4 - off
17.	GASMAP Calibration Module, Front Panel Disconnect the following: • GASMAP Calibration Module (sample output) from Catheter NOTE: The catheter is disconnected to relieve pressure on the unit.	
18.	GASMAP Calibration Module, Front Panel Connect the following: • GASMAP Calibration Module (sample output) to Catheter	
19.	External Low Pressure Gauge, Front Panel Verify the following: • Gauge reading is < 1 psi	
20.		GSE Calibration Solenoid Test Box • SW1 - off • SW2 - on • SW3 - off • SW4 - off Verify the following: • SOL2 LED is illuminated • Audible click can be heard NOTE: Allow fifteen (15) seconds for the pressure to stabilize before continuing with the following step.
21.	External Low Pressure Gauge, Front Panel Record the following information: • Tank 2 pressure: _____ psi ($6 \leq P \leq 10$)	
22.		GSE Calibration Solenoid Test Box • SW1 - off • SW2 - off • SW3 - off • SW4 - off
23.	GASMAP Calibration Module, Front Panel Disconnect the following: • GASMAP Calibration Module (sample output) from Catheter NOTE: The catheter is disconnected in order to relieve pressure on the unit. When the pressure has been relieved, continue on with the following step.	
24.	GASMAP Calibration Module, Front Panel Connect the following: • GASMAP Calibration Module (sample output) to Catheter	
25.	External Low Pressure Gauge, Front Panel Verify the following: • Gauge reading is < 1 psi	

TABLE 6.22 GASMAP CALIBRATION FUNCTIONAL (CONT'D)

Step	Payload	Tech																
26.		GSE Calibration Solenoid Test Box <ul style="list-style-type: none">• SW1 - off• SW2 - off• SW3 - on• SW4 - off Verify the following: <ul style="list-style-type: none">• SOL3 LED is illuminated• Audible click can be heard NOTE: Allow fifteen (15) seconds for the pressure to stabilize before continuing with the following step.																
27.	External Low Pressure Gauge, Front Panel Record the following information: <ul style="list-style-type: none">• Tank 3 pressure: _____ psi ($6 \leq P \leq 9$)																	
28.		GSE Calibration Solenoid Test Box <ul style="list-style-type: none">• SW1 - off• SW2 - off• SW3 - off• SW4 - off																
29.	GASMAP Calibration Module, Front Panel Disconnect the following: <ul style="list-style-type: none">• GASMAP Calibration Module (sample output) from Catheter NOTE: The catheter is disconnected to relieve pressure on the unit.																	
30.	GASMAP Calibration Module, Front Panel Connect the following: <ul style="list-style-type: none">• GASMAP Calibration Module (sample output) to Catheter																	
31.	External Low Pressure Gauge, Front Panel Verify the following: <ul style="list-style-type: none">• Gauge reading is < 1 psi																	
32.		Variable Power Supply Main pwr sw - off																
33.		Disconnect the following hardware: <ul style="list-style-type: none">• Variable Power Supply from Power Source• GSE Calibration Solenoid Test Box from Variable Power Supply																
34.	GASMAP Calibration Module, Front Panel Disconnect the following: <ul style="list-style-type: none">• GSE Calibration Solenoid Test Box, P1 from GASMAP• Calibration Module, J1 (SOL POWER)																	
36.	GASMAP Calibration Module, Front Panel <ul style="list-style-type: none">• Valve 1 - closed• Valve 2 - closed• Valve 3 - closed																	
37.	Disassemble all GASMAP hardware Return the following hardware to Building 241 Controlled Storage: <table><tr><th>Qty</th><th>Item</th></tr><tr><td>1</td><td>GASMAP Calibration Module</td></tr><tr><td>1</td><td>Catheter</td></tr><tr><td>1</td><td>GSE Calibration Solenoid Test Box</td></tr><tr><td>1</td><td>External Low Pressure Gauge</td></tr><tr><td>1</td><td>Power Cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></table>	Qty	Item	1	GASMAP Calibration Module	1	Catheter	1	GSE Calibration Solenoid Test Box	1	External Low Pressure Gauge	1	Power Cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item																	
1	GASMAP Calibration Module																	
1	Catheter																	
1	GSE Calibration Solenoid Test Box																	
1	External Low Pressure Gauge																	
1	Power Cable																	
1	Variable Power Supply																	
1	Digital Multimeter																	

6.3.1.3 Clearing Errors

The following sequence is to be performed when the “ERROR” LED is illuminated on the GASMAP payload. This sequence can be performed during the payload initial power up procedure, or can be executed at any time in Section 6.3.1.1 when the “ERROR” LED is illuminated. If an “LOV” error exists on the GASMAP error log, this sequence MUST be followed by Analyzer Roughing in Section 6.3.1.4. If an “LOV” error does not exist, clear the errors and proceed to Section 6.3.1.1.

TABLE 6.23 CLEARING ERRORS

Step	PRCU	HRF Rack 1
1.	GASMAP Analyzer Module, “Main Menu” Screen Select: “Low Pwr” <5> Select: “Yes” <ENTER> Select: “Diag” <3> Select: “Display” <1> Select: “Errors” <4> Select: “Error Log” <1> Verify the following is displayed: <ul style="list-style-type: none">• “Diagnostic” Screen	
2.	GASMAP Analyzer Module, “Diagnostic” Screen Verify the following: <ul style="list-style-type: none">• The number of error entries currently recorded is displayed. Select: <1> For each error entry, perform the following: <ul style="list-style-type: none">• Log Error in test log Select: “Retriever Error Code” <1> Select: “Acknowledge Error Code” <2> Select: “Delete Error Code” <3> Select: “Yes” <ENTER> Verify the following: <ul style="list-style-type: none">• All errors have been deleted NOTE: If LED is still illuminated, scroll down to look at the next entry in the file. Repeat this step on all other error entries. Select: <MAIN MENU> Verify the following is displayed: <ul style="list-style-type: none">• “Main Menu” Screen NOTE: If the “LOV” error occurred, perform the Section 6.3.1.4	

6.3.1.4 Analyzer Roughing

This process is to be performed if an “LOV” error occurs during the execution of Section 6.3.1.3. Once the roughing has been completed, the GASMAP unit must be powered off. This may require the test engineer to repeat the complete GASMAP functional in Section 6.3.1.1.

TABLE 6.24 ANALYZER ROUGHING

Step	PRCU	HRF Rack 1
1.	GASMAP Analyzer Module, "Main Menu" Screen Select: "Roughing"	
2.	Connect the following hardware: <ul style="list-style-type: none"> Roughing pump to GASMAP Analyzer Module (Roughing Port) 	
3.	Roughing pump, Main pwr sw - on Following instructions on Screen Select: <MAIN MENU>	

6.3.2 Cooling Stowage Drawer

This section describes the procedures for the functional test of the Cooling Stowage Drawers. The following table describes the flight units or equivalent hardware that may be used to perform off-line functional tests.

TABLE 6.25 COOLING STOWAGE DRAWER HARDWARE

Qty	Item	Class
1	Cooling Stowage Drawer	Flight

The objective of this test is to assess the functionality of the Cooling Stowage Drawer to ensure the hardware is functioning properly before being integrated into the HRF Rack.

TABLE 6.26 COOLING STOWAGE DRAWER

Step	Payload	GSE Tech																		
1.	<p>Remove the following hardware from Building 241 Controlled Storage:</p> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>Cooling Stowage Drawer</td></tr><tr><td>1</td><td>28 Vdc pwr cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Volt Meter</td></tr><tr><td>1</td><td>Data Cable (Optional)</td></tr><tr><td>1</td><td>Ground Station Control (GSC) Power Cable (Optional)</td></tr><tr><td>1</td><td>Generic Breakout Box (Optional)</td></tr><tr><td>1</td><td>Switch (Optional)</td></tr></tbody></table> <p>NOTE: Optional hardware is only needed for detailed functional checkout.</p>	Qty	Item	1	Cooling Stowage Drawer	1	28 Vdc pwr cable	1	Variable Power Supply	1	Digital Volt Meter	1	Data Cable (Optional)	1	Ground Station Control (GSC) Power Cable (Optional)	1	Generic Breakout Box (Optional)	1	Switch (Optional)	
Qty	Item																			
1	Cooling Stowage Drawer																			
1	28 Vdc pwr cable																			
1	Variable Power Supply																			
1	Digital Volt Meter																			
1	Data Cable (Optional)																			
1	Ground Station Control (GSC) Power Cable (Optional)																			
1	Generic Breakout Box (Optional)																			
1	Switch (Optional)																			

TABLE 6.26 COOLING STOWAGE DRAWER (CONT'D)

Step	Payload	GSE Tech
2.		<p>Variable Power Supply</p> <p>Verify the following:</p> <p>Main pwr sw - off</p> <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p> <p>N/A: ____ T: ____ QA: ____</p> <p>NOTE: The following three steps are to perform a detailed checkout of the data connections.</p>
3.		<p>Connect the following hardware:</p> <ul style="list-style-type: none"> Data Cable SED38115055-301 to Break Out Box
4.	<p>Connect the following hardware:</p> <ul style="list-style-type: none"> Data Cable SED38115055-301 to Cooling Stowage Drawer 	
5.		<p>Digital Volt Meter</p> <p>Verify the following:</p> <ul style="list-style-type: none"> 1 amp short between pins 9 and 121 <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p> <p>N/A: ____ T: ____ QA: ____</p> <p>(Previous three (3) steps)</p>
6.		<p>Connect the following hardware:</p> <ul style="list-style-type: none"> 28 V dc pwr cable (GSE) to Variable Power Supply Variable Power Supply to Power Source
7.		<p>Variable Power Supply, front panel</p> <p>Main Power sw - on</p> <p>Set Value: 28 V</p> <p>Set Value: 4 amps</p> <p>NOTE: If using a calibrated power supply, continue on with the following step:</p> <p>Using Digital Volt Meter:</p> <ul style="list-style-type: none"> Verify the voltage between pins 1 and 2 measures 28 ± 0.5 V dc <p>Record the following information:</p> <p>Model: _____</p> <p>NASA Tag #: _____</p> <p>Calibration #: _____</p> <p>Calibration Date: _____</p>
8.		<p>Variable Power Supply</p> <p>Main pwr sw - off</p>
9.	<p>Connect the following hardware:</p> <ul style="list-style-type: none"> 28 V dc pwr cable (GSE) to Cooling Stowage Drawer (J1) 	
10.		<p>Variable Power Supply, front panel</p> <p>Main Power sw - on</p> <p>Set Value: 28 V</p> <p>Set Value: 4 amps</p>
11.	<p>Cooling Stowage Drawer, Rear panel</p> <p>Verify the following:</p> <ul style="list-style-type: none"> Fan audibly engages Air exhaust is exiting payload 	

TABLE 6.26 COOLING STOWAGE DRAWER (CONT'D)

Step	Payload	GSE Tech										
12.		Digital Volt Meter Record the following: <ul style="list-style-type: none">Voltage between pins 24 & 44: _____ V dc (1.8 ± 0.6 V dc) <div>N/A: _____ T: _____ QA: _____</div>										
13.		Variable Power Supply Main pwr sw <div>- off</div>										
14.		Disconnect the following hardware: <ul style="list-style-type: none">Variable Power Supply from Power Source28 V dc pwr cable (GSE) from Cooling Stowage Drawer (J1)28 V dc pwr cable (GSE) from Variable Power Supply										
15.	Disassemble Cooling Stowage Drawer hardware Return the following hardware from Building 241 Controlled Storage: <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>2</td><td>Cooling Stowage Drawer</td></tr><tr><td>1</td><td>28 Vdc pwr cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></tbody></table>	Qty	Item	2	Cooling Stowage Drawer	1	28 Vdc pwr cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item											
2	Cooling Stowage Drawer											
1	28 Vdc pwr cable											
1	Variable Power Supply											
1	Digital Multimeter											

T: _____ QA: _____

6.3.3 Ultrasound

The Ultrasound functional test confirms the Ultrasound System is working properly and focuses on the modified COTS ultrasound components. The test accurately assesses eighty-five percent (85%) of the system and will confirm the functionality of those systems and the following subsystems:

- System Software
- Critical Data Pathways
- Video Tape Recorder Power/Power Supply Board
- Fans/Tachometer
- Direct Current (DC) Interface Module
- DC/DC Converter (Primary)
- Front Panel LED

The following list provides the flight units or equivalent hardware which may be used to perform off line functional tests.

TABLE 6.27 ULTRASOUND HARDWARE

Qty	Item	Class
1	Ultrasound Unit Assy	Flight
1	Keyboard Module Assy	Flight
1	Keyboard Cable Assy	Flight
1	External Monitor Cable Assy	Flight
1	Monitor Cable Assy	Flight
1	HRF Monitor	Flight
1	Transducer	Flight

Once the Ultrasound payload has successfully completed functional testing, it can be integrated into the HRF Rack.

TABLE 6.28 ULTRASOUND FUNCTIONAL CHECKOUT

Step	Payload	GSE Tech																						
1.	Remove the following hardware from Building 241 Controlled Storage: <table><tr><td>Qty</td><td>Item</td></tr><tr><td>1</td><td>Ultrasound Unit Assembly</td></tr><tr><td>1</td><td>Keyboard Module Assy</td></tr><tr><td>1</td><td>Keyboard Cable Assy</td></tr><tr><td>1</td><td>Transducer</td></tr><tr><td>1</td><td>Monitor Cable Assy</td></tr><tr><td>1</td><td>GSE Power Cable</td></tr><tr><td>1</td><td>HRF Monitor</td></tr><tr><td>1</td><td>Ultrasound 28 Vdc pwr cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></table>	Qty	Item	1	Ultrasound Unit Assembly	1	Keyboard Module Assy	1	Keyboard Cable Assy	1	Transducer	1	Monitor Cable Assy	1	GSE Power Cable	1	HRF Monitor	1	Ultrasound 28 Vdc pwr cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item																							
1	Ultrasound Unit Assembly																							
1	Keyboard Module Assy																							
1	Keyboard Cable Assy																							
1	Transducer																							
1	Monitor Cable Assy																							
1	GSE Power Cable																							
1	HRF Monitor																							
1	Ultrasound 28 Vdc pwr cable																							
1	Variable Power Supply																							
1	Digital Multimeter																							
2.	Assemble the hardware from above per Appendix B Figure 2																							
3.	HRF Wrkstn, Front Panel Verify the following: <ul style="list-style-type: none">• “VTR Power” sw - off• “Ultrasound Power” sw - off																							
4.	Keyboard Module Assy, Left Side Panel Verify the following: “Standby Power” sw - STBY																							
5.	HRF Monitor Verify the following: <ul style="list-style-type: none">• Main pwr sw - off																							
6.		Variable Power Supply Verify the following: <ul style="list-style-type: none">• Main pwr sw - off Record the following information: Model: _____ NASA Tag #: _____ Calibration #: _____ Calibration Date: _____ <div>N/A: _____ T: _____ QA: _____</div>																						
7.		Connect the following hardware: <ul style="list-style-type: none">• Ultrasound 28 V dc pwr cable (GSE) to Variable Power Supply• Variable Power Supply to Power Source																						

TABLE 6.28 ULTRASOUND FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	GSE Tech
8.		Variable Power Supply, front panel Main Power sw - on Set Value: 28 V Set Value: 40 amps NOTE: If using a calibrated power supply, continue on with the following step. Using Digital Multimeter: <ul style="list-style-type: none"> Verify the voltage between pins 1 and 2 measures 28 ± 0.5 V dc Record the following information: Model: _____ NASA Tag #: _____ Calibration #: _____ Calibration Date: _____
9.		Variable Power Supply Main pwr sw - off
10.	Connect the following hardware: <ul style="list-style-type: none"> Ultrasound 28 V dc pwr cable (GSE) to Ultrasound Unit 	
11.		Variable Power Supply, front panel Main Power sw - on Set Value: 28 V Set Value: 40 amps
12.	HRF Monitor, Right Side Panel Function sw - Ultrasound Main pwr sw - on	
13.	Ultrasound Unit Assy., front panel "Ultrasound Power" sw - on NOTE: Wait fifteen (15) seconds before proceeding.	
14.	Keyboard Module Assy, Right Side Panel "Standby Power" sw - on	
15.	Ultrasound Unit Assy., front panel Verify the following: <ul style="list-style-type: none"> Ultrasound Power LED is on 	
16.	HRF Monitor, front panel Verify the following appears: <ul style="list-style-type: none"> Sonogram Screen Operational Clock NOTE: If warning or error banners appear, record them in the error log by pressing the <Superkey><THI><0> two (2) times before proceeding with the following step. Remove scanhead from the Ultrasound	
17.	Keyboard Module Assy, Top Panel Select: <Superkey><0> Verify the following appears: <ul style="list-style-type: none"> Ultrasound Login display 	
18.	Keyboard Module Assy, Top Panel At the "User" prompt: Type: csr<Enter> At the "Password" prompt: Type: user<Enter> Select: "OK" button	
19.	Keyboard Module Assy "Tests and Utilities" Screen Select: FREEZE Select: "Test, Utils" button Verify the following appears: <ul style="list-style-type: none"> "Machine Components" Screen 	

TABLE 6.28 ULTRASOUND FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	GSE Tech
20.	Keyboard Module Assy "Machine Components" Screen Select: "Machine..." Verify the following: <ul style="list-style-type: none"> "2D Ultrasound" is highlighted 	
21.	Keyboard Module Assy "Machine Components" Screen Select: "Tests & Utils." Verify the following: <ul style="list-style-type: none"> "Test and Utilities" Screen appears 	
22.	Keyboard Module Assy "Test and Utilities" Screen Select: End-to-End Test Select: "Execute" button Verify the following: <ul style="list-style-type: none"> Test indicates PASS 	
23.	Keyboard Module Assy "Test and Utilities" Screen Select: Back-End Test Select: "Execute" button <ul style="list-style-type: none"> Wait approximately two (2) minutes Verify the following: <ul style="list-style-type: none"> Test indicates PASS 	
24.	Keyboard Module Assy "Test and Utilities" Screen Select: "Components" button Verify the following: <ul style="list-style-type: none"> "Machine Components" Screen appears 	
25.	Keyboard Module Assy "Machine Components" Screen Select: "Color Ultrasound" Select: "Test & Utils." Button Verify the following: <ul style="list-style-type: none"> "Test and Utilities" Screen appears 	
26.	Keyboard Module Assy "Test and Utilities" Screen Select: End-to-End Test Select: "Execute" button <ul style="list-style-type: none"> Wait approximately two (2) minutes Verify the following: <ul style="list-style-type: none"> Test indicates PASS 	
27.	Keyboard Module Assy "Tests and Utilities" Screen Select: Back-End Test Select: "Execute" button Verify the following: <ul style="list-style-type: none"> Test indicates PASS 	
28.	Keyboard Module Assy "Tests and Utilities" Screen Select: "Components" Select: "Go to Top Level..." button Select: "Machine..." Select: "Test & Utils." Button Verify the following: <ul style="list-style-type: none"> "Test and Utilities" screen appears 	
29.	Keyboard Module Assy "Test and Utilities" Screen Select: "Comprehensive Test" Select: "Execute" button NOTE: Execution of this test takes approximately twenty (20) minutes. Verify the following: <ul style="list-style-type: none"> Test indicates PASS 	
30.	Keyboard Module Assy "Test and Utilities" Screen Select: "Close" Verify the following: <ul style="list-style-type: none"> "ATL" window appears 	

TABLE 6.28 ULTRASOUND FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	GSE Tech																		
31.	Reattach Scanhead to Ultrasound Unit NOTE: Reattach Scanhead by keeping the handle in an upright position and slide Scanhead back to connect to the Ultrasound unit. Lock in place by rotating the handle to the right.																			
32.	Keyboard Module Assy "ATL" Window Select: "REBOOT"																			
33.	HRF Monitor, front panel Verify the following appears: <ul style="list-style-type: none">• Sonogram Screen• Operational Clock NOTE: If warning or error banners appear, record them in the error log by pressing the <Superkey><THI><0> two (2) times.																			
34.	Keyboard Module Assy, Left Side Panel "Main" sw - STBY																			
35.	Ultrasound Unit Assy., front panel Verify the following: <ul style="list-style-type: none">• Ultrasound Power LED is off																			
36.	HRF Monitor, Front Panel Verify the following: <ul style="list-style-type: none">• Display disappears																			
37.	Ultrasound Unit Assy., front panel "Ultrasound Power" sw - off																			
38.	HRF Monitor, Panel pwr - off Main Power sw - off																			
39.		Variable Power Supply, front panel Main Power sw - off																		
40.		Disconnect the following hardware: <ul style="list-style-type: none">• Variable Power Supply from Power Source• Ultrasound 28 V dc pwr cable (GSE) from HRF Wrkstn (J1)• Ultrasound 28 V dc pwr cable (GSE) from Variable Power Supply																		
41.	Disassemble All Ultrasound hardware Return the following items to Building 241 Controlled Storage <table><tr><td><u>Qty</u></td><td><u>Item</u></td></tr><tr><td>1</td><td>Ultrasound Unit Assy.</td></tr><tr><td>1</td><td>Keyboard Module Assy</td></tr><tr><td>1</td><td>Keyboard Cable Assy</td></tr><tr><td>1</td><td>Monitor Cable Assy</td></tr><tr><td>1</td><td>HRF Monitor</td></tr><tr><td>1</td><td>Ultrasound 28 Vdc pwr cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></table>	<u>Qty</u>	<u>Item</u>	1	Ultrasound Unit Assy.	1	Keyboard Module Assy	1	Keyboard Cable Assy	1	Monitor Cable Assy	1	HRF Monitor	1	Ultrasound 28 Vdc pwr cable	1	Variable Power Supply	1	Digital Multimeter	
<u>Qty</u>	<u>Item</u>																			
1	Ultrasound Unit Assy.																			
1	Keyboard Module Assy																			
1	Keyboard Cable Assy																			
1	Monitor Cable Assy																			
1	HRF Monitor																			
1	Ultrasound 28 Vdc pwr cable																			
1	Variable Power Supply																			
1	Digital Multimeter																			

T: _____ QA: _____

6.3.4 Workstation

This section describes the procedures for the functional test of the Workstation payload drawer.

The following table describes the flight units or equivalent hardware that may be used to perform off-line functional tests.

TABLE 6.29 WORKSTATION HARDWARE

Qty	Item	Class
1	HRF Workstation	Flight
1	HRF Workstation Keyboard	Flight
1	HRF Monitor	Flight
1	Workstation Monitor Cable	Flight
1	Workstation (WS) Keyboard/Mouse cable	Flight
1	Serial cable	Flight

Once the Workstation payload has successfully completed functional testing, it can be integrated into the HRF Rack.

TABLE 6.30 WORKSTATION FUNCTIONAL

Step	Payload	GSE Tech																				
1.	<div>Remove the following hardware from Building 241 Controlled Storage:</div> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>HRF Wrkstn</td></tr><tr><td>1</td><td>HRF Wrkstn keyboard</td></tr><tr><td>1</td><td>HRF Monitor</td></tr><tr><td>1</td><td>Wrkstn Monitor cable</td></tr><tr><td>1</td><td>WS Keyboard/Mouse cable</td></tr><tr><td>1</td><td>Serial cable</td></tr><tr><td>1</td><td>28 Vdc cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></tbody></table>	Qty	Item	1	HRF Wrkstn	1	HRF Wrkstn keyboard	1	HRF Monitor	1	Wrkstn Monitor cable	1	WS Keyboard/Mouse cable	1	Serial cable	1	28 Vdc cable	1	Variable Power Supply	1	Digital Multimeter	
Qty	Item																					
1	HRF Wrkstn																					
1	HRF Wrkstn keyboard																					
1	HRF Monitor																					
1	Wrkstn Monitor cable																					
1	WS Keyboard/Mouse cable																					
1	Serial cable																					
1	28 Vdc cable																					
1	Variable Power Supply																					
1	Digital Multimeter																					
2.	<div>HRF Wrkstn, Front Panel</div> <div>Verify the following:</div> <ul style="list-style-type: none">• “Main Power” sw - off																					
3.	<div>HRF Monitor</div> <div>Verify the following:</div> <ul style="list-style-type: none">• Main pwr sw - off																					
4.	<div>Assemble the hardware from above per Figure B-3 Connections for WS Testing</div>																					
5.		<div>Variable Power Supply</div> <div>Verify the following:</div> <ul style="list-style-type: none">• Main pwr sw - off <div>Record the following information:</div> <div>Model: _____</div> <div>NASA Tag #: _____</div> <div>Calibration #: _____</div> <div>Calibration Date: _____</div> <div>N/A: _____ T: _____ QA: _____</div>																				

TABLE 6.30 WORKSTATION FUNCTIONAL (CONT'D)

Step	Payload	GSE Tech
6.		Connect the following hardware: <ul style="list-style-type: none"> • 28 V dc pwr cable (GSE) to Variable Power Supply • Variable Power Supply to Power Source
7.		Variable Power Supply, front panel Main Power sw - on Set Value: 28 V Set Value: 10 amps NOTE: If using a calibrated power supply, continue on with the following step. Using Digital Multimeter: <ul style="list-style-type: none"> • Verify the voltage between pins 1 and 2 measures 28 ± 0.5 V dc Record the following information: Model: _____ NASA Tag #: _____ Calibration #: _____ Calibration Date: _____
8.		Variable Power Supply <ul style="list-style-type: none"> • Main pwr sw - off
9.		Connect the following hardware: <ul style="list-style-type: none"> • 28 V dc pwr cable (GSE) to HRF Wrkstn (J1)
10.		Variable Power Supply, front panel Main Power sw - on Set Value: 28 V Set Value: 10 amps
11.	HRF Wrkstn, Front Panel "Main Power" sw - on Verify the following: <ul style="list-style-type: none"> • Main Power LED is illuminated green 	
12.	HRF Monitor, Front Panel Function sw - Wrkstn "Main Power" sw - on Verify the following: <ul style="list-style-type: none"> • Main Power LED is illuminated green 	
13.	HRF Monitor Select: "Window NT 4.0" <ENTER> NOTE: Wrkstn will automatically boot into Windows NT 4.0 after thirty (30) seconds if no user input is provided. Verify the following is displayed: <ul style="list-style-type: none"> • "Press Ctrl + Alt + Delete to log on" Dialog box 	
14.	HRF Wrkstn, "Press Ctrl + Alt + ..." Dialog box Select: <Ctrl><Alt><Delete> Verify the following is displayed: <ul style="list-style-type: none"> • "Login Information" Window 	
15.	HRF Wrkstn, "Login Information" Window AT "user" prompt: Type: "administrator" At "password" prompt: Select: "hrf" Verify the following: <ul style="list-style-type: none"> • "Windows NT" desktop appears 	
16.	HRF Wrkstn, "Windows NT" desktop Select: "Start" menu Navigate to: "CSW User Interface" Verify the following: <ul style="list-style-type: none"> • "HRF" screen appears 	

TABLE 6.30 WORKSTATION FUNCTIONAL (CONT'D)

Step	Payload	GSE Tech
17.	HRF Wrkstn, "HRF" Screen Select: "HRF Exit" button Verify the following: • Dialog box appears	
18.	HRF Wrkstn, dialog box Select: "Yes" button Verify the following: • "Windows NT" desktop appears	
19.	HRF Wrkstn, "Windows NT" desktop Select: "Start" menu Select: "Programs" Select: "Wrkstn" Verify the following: • "Wrkstn" window appears	
20.	HRF Wrkstn, "Wrkstn" Window Select: "Individual tests" button Select: "Graphics test" button Verify the following: • Dialog box appears NOTE: Execution of this test takes approximately twenty (20) minutes.	
21.	HRF Wrkstn, Dialog box Select: "Yes" button	
22.	HRF Wrkstn, "Wrkstn" Window Verify the following: • "Graphics Test: "OK" message appears	
23.	HRF Wrkstn, "Wrkstn" Window Select: "Exit" button Verify the following: • Dialog box appears	
24.	HRF Wrkstn, Dialog box Select: "Yes" button Verify the following: • "Windows NT" desktop appears	
25.	HRF Wrkstn, "Windows" desktop Select: "Start" menu Select: "Shutdown" Verify the following is displayed: • "Shut Down Windows" Window	
26.	HRF Wrkstn, "Shut Down Windows" Window Select: "YES" button Verify the following is displayed: • "It is now safe to turn off your computer" message	
27.	HRF Wrkstn, Front Panel "Main Power" sw - off Verify the following: • "Main Power" LED is not illuminated	
28.	HRF Monitor Panel Pwr - off Main pwr sw - off Verify the following: • "Main Power" LED is not illuminated	
29.	HRF Wrkstn, Front Panel 28 V dc pwr sw - off Verify the following: • 28 V dc LED is not illuminated	
30.		Variable Power Supply Main pwr sw - off
31.		Disconnect the following hardware: • Variable Power Supply from Power Source • 28 V dc pwr cable (GSE) from HRF Wrkstn (J1) • 28 V dc pwr cable (GSE) from Variable Power Supply

TABLE 6.30 WORKSTATION FUNCTIONAL (CONT'D)

Step	Payload	GSE Tech																				
32.	<p>Disassemble Wrkstn hardware</p> <p>Return the following hardware from Building 241 Controlled Storage:</p> <table><tr><th>Qty</th><th>Item</th></tr><tr><td>1</td><td>HRF Wrkstn</td></tr><tr><td>1</td><td>HRF Wrkstn keyboard</td></tr><tr><td>1</td><td>HRF Monitor</td></tr><tr><td>1</td><td>Wrkstn Monitor cable</td></tr><tr><td>1</td><td>WS Keyboard/Mouse cable</td></tr><tr><td>1</td><td>Serial cable</td></tr><tr><td>1</td><td>28 Vdc cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr></table>	Qty	Item	1	HRF Wrkstn	1	HRF Wrkstn keyboard	1	HRF Monitor	1	Wrkstn Monitor cable	1	WS Keyboard/Mouse cable	1	Serial cable	1	28 Vdc cable	1	Variable Power Supply	1	Digital Multimeter	
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1	Wrkstn Monitor cable																					
1	WS Keyboard/Mouse cable																					
1	Serial cable																					
1	28 Vdc cable																					
1	Variable Power Supply																					
1	Digital Multimeter																					

T: _____ QA: _____

6.3.5 Test Support System Functional

The following list provides the hardware that may be used to perform functional tests.

TABLE 6.31 TEST SUPPORT SYSTEM HARDWARE

Qty	Item	Class
1	Test Support System	GSE
1	Signal Adapter Box	GSE
1	SAB/Laptop RS232 Cable	GSE
1	SAB/Laptop Video Cable	GSE
2	SAB/BOB Interface Cable	GSE
1	Ethernet Cable	GSE
2	DAQCARD Interface	GSE
2	Portable Computer Memory Card International Adapter (PCMCIA) Card	GSE
1	Interface Cable	GSE
1	Interface Cable	GSE
1	IBM Video Adapter Cable	GSE
1	Ethernet PCMCIA Card	
2	BOB	GSE

Once the Test Support System has successfully completed functional testing, it is considered operational and can be used during testing of the HRF Rack.

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT

Step	Payload	Tech Functional Checkout																																		
1.	<div>Remove the following hardware from Building 241 Controlled Storage:</div> <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>TSS Laptop</td></tr><tr><td>2</td><td>DAQCARD</td></tr><tr><td>1</td><td>PA50-50F</td></tr><tr><td>2</td><td>DAQ 1200/SAB interface cable</td></tr><tr><td>1</td><td>SAB</td></tr><tr><td>1</td><td>Ethernet PCMCIA card</td></tr><tr><td>1</td><td>3C-PC-COMBO-Commercial Bills of Lading (CBL)</td></tr><tr><td>1</td><td>Ethernet cable</td></tr><tr><td>1</td><td>Video Adapter cable</td></tr><tr><td>1</td><td>SAB/laptop Video cable</td></tr><tr><td>1</td><td>SAB/Laptop RS-232 cable</td></tr><tr><td>2</td><td>BOB</td></tr><tr><td>1</td><td>GSE Power cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr><tr><td>2</td><td>SAB/BOB Interface Cable</td></tr></tbody></table>	Qty	Item	1	TSS Laptop	2	DAQCARD	1	PA50-50F	2	DAQ 1200/SAB interface cable	1	SAB	1	Ethernet PCMCIA card	1	3C-PC-COMBO-Commercial Bills of Lading (CBL)	1	Ethernet cable	1	Video Adapter cable	1	SAB/laptop Video cable	1	SAB/Laptop RS-232 cable	2	BOB	1	GSE Power cable	1	Variable Power Supply	1	Digital Multimeter	2	SAB/BOB Interface Cable	
Qty	Item																																			
1	TSS Laptop																																			
2	DAQCARD																																			
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1	Ethernet cable																																			
1	Video Adapter cable																																			
1	SAB/laptop Video cable																																			
1	SAB/Laptop RS-232 cable																																			
2	BOB																																			
1	GSE Power cable																																			
1	Variable Power Supply																																			
1	Digital Multimeter																																			
2	SAB/BOB Interface Cable																																			
2.	Assemble hardware per Appendix B Figure 7 & 8																																			
3.	BOB1 front panel "DATA" sw - off																																			
4.	BOB2 front panel DATA" sw - off																																			
5.	TSS Laptop pwr - off																																			
6.		<div>Variable Power Supply</div> <div>Verify the following:</div> <div>Main pwr sw - off</div> <div>Record the following information:</div> <div>Model: _____</div> <div>NASA Tag #: _____</div> <div>Calibration #: _____</div> <div>Calibration Date: _____</div> <div>N/A: ____ T: ____ QA: ____</div>																																		
7.		<div>Connect the following hardware:</div> <div><ul style="list-style-type: none">GSE Power Cable to Variable Power SupplyVariable Power Supply to Power Source</div>																																		
8.		<div>Variable Power Supply, front panel</div> <div>Main Power sw - on</div> <div>Set Value: 28 V</div> <div>Set Value: 25 amps</div> <div>NOTE: If using a calibrated power supply, continue on with the following step. If powering two (2) BOB simultaneously, set amperage to fifty (50) amps.</div> <div>Using Digital Multimeter:</div> <div><ul style="list-style-type: none">Verify the voltage between pins 1 and 2 measures 28 ± 0.5 V dc</div> <div>Record the following information:</div> <div>Model: _____</div> <div>NASA Tag #: _____</div> <div>Calibration #: _____</div> <div>Calibration Date: _____</div>																																		

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech Functional Checkout
9.		Variable Power Supply Main pwr sw - off
10.		Connect the following hardware: <ul style="list-style-type: none"> • BOB1 to Variable Power Supply • BOB2 to Variable Power Supply
11.		Variable Power Supply, front panel Main Power sw - on Set Value: 28 V Set Value: 25 amps
12.	TSS laptop left side panel, Main pwr sw - on	
13.	TSS laptop display, Verify the following: <ul style="list-style-type: none"> • "Windows 95" executes 	
14.	TSS laptop display "Windows NT" desktop Select: "Start" menu Select: "Programs" icon Select: "National Instruments DAQ" icon Select: "NI-DAQ" test panels Verify the following is displayed: <ul style="list-style-type: none"> • "Choose a Device" Window 	
15.	TSS laptop display "Choose a Device" Window Select: "Device #1 DAQCARD 1200" Select: "OK" button Verify the following is displayed: <ul style="list-style-type: none"> • "Test Panel" Window 	
16.	TSS laptop display "Test Panel" Window Verify "Fatal Error" light is illuminated gray Select: "CLOSE" button Verify the following is displayed: <ul style="list-style-type: none"> • "Windows 95" desktop 	
17.	TSS laptop display "Windows NT" desktop Select: "Start" menu Select: "Programs" icon Select: "National Instruments DAQ" icon Select: "NI-DAQ" test panels Verify the following is displayed: <ul style="list-style-type: none"> • "Choose a Device" Window 	
18.	TSS laptop display "Choose a Device" Window Select: "Device #2 DAQCARD 1200" Select: "OK" button Verify the following is displayed: <ul style="list-style-type: none"> • "Test Panel" Window 	
19.	TSS laptop display "Test Panel" Window Verify the following: <ul style="list-style-type: none"> • "Fatal Error" light is illuminated gray Select: "CLOSE" button Verify the following is displayed: <ul style="list-style-type: none"> • "Windows 95" desktop 	
20.	TSS laptop display "Windows 95" desktop Select: "Start" menu Select: "Settings" icon Select: "Control Panels" icon Select: "Network" icon Select: "TCP/IP -> 3 COM ..." command Verify the following is displayed: <ul style="list-style-type: none"> • "Network" Window 	
21.	TSS laptop display "Network" Window <ul style="list-style-type: none"> • Record IP address: _____ Select: "OK" button Select: "OK" button Verify the following is displayed: <ul style="list-style-type: none"> • "Control Panel" Window 	

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech Functional Checkout
22.	TSS laptop display "Control Panel" Window Select: "File" menu Select: "Close" Verify the following is displayed: • "Windows 95" desktop	
23.	TSS laptop display "Windows 95" desktop Select: "Start" menu Select: "Programs" icon Select: "MS-DOS Prompt" Verify the following is displayed: • "MS-DOS" Window	
24.	TSS laptop display "MS-DOS" Window Verify the following is displayed: • "c:\Windows>" prompt	
25.	TSS laptop • Type: Ping [Insert IP address from step 21.] <Enter>	
26.	TSS laptop display "MS-DOS" Window Verify the following is displayed: • "Reply" response	
27.	TSS laptop • Type: exit <Enter> Verify the following is displayed: • "Windows 95" desktop	
28.	TSS laptop display "Windows 95" desktop Select: "START" button Select: "I&ES Simulator (Start).exe" icon Verify the following is displayed: • "I&ES Configuration" Window	
29.	TSS laptop display "I&ES Configuration" Window Select: "START" button NOTE: Wait fifteen (15) seconds for boot up of I&ES software. Verify the following is displayed: • "I&ES Simulator 1" Window • "I&ES Simulator 2" Window	
30.	TSS laptop display "I&ES Simulator 1" Window Select: Stop Simulator 1 "QUIT" button	
31.	TSS laptop display "I&ES Simulator 1" Window Select: Stop Simulator 2 "QUIT" button Verify the following is displayed: • "Windows 95" desktop	
32.	TSS laptop display "Windows 95" desktop Select: "Start" menu Select: "Shutdown" icon Verify the following is displayed: • "Shutdown" Window	
33.	TSS laptop display "Shutdown" Window Select: "YES" button	
34.		BOB1 front panel Verify the following: 2.5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
35.		BOB1 front panel Data sw - on Verify data LED - on Fan sw - on Verify the following: • 2.5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off
36.		BOB1 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on
37.		BOB1 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - off
38.		BOB1 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - off • 5 amp LED - off
39.		BOB1 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
40.		BOB1 front panel 2.5 amp sw - on 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off
41.		BOB1 front panel 2.5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off NOTE: After continuous operation of loads in excess of five (5) minutes, the fan must be allowed to run for ten (10) minutes.
42.		BOB1 front panel Data sw - off Verify data LED - off Fan sw - off
43.		BOB2 front panel Verify the following: 2.5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off
44.		BOB2 front panel Data sw - on Verify data LED - on Fan sw - on Verify the following: • 2.5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off
45.		BOB2 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech
46.		BOB2 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - off
47.		BOB2 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - on 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - on • 5 amp LED - off • 5 amp LED - off
48.		BOB2 front panel 2.5 amp sw - on 5 amp sw - on 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - on • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off
49.		BOB2 front panel 2.5 amp sw - on 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - on • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off
50.		BOB2 front panel 2.5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off 5 amp sw - off Verify the following: • 2.5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off • 5 amp LED - off NOTE: After continuous operation of loads in excess of five (5) minutes, the fan must be allowed to run for ten (10) minutes.

TABLE 6.32 TEST SUPPORT SYSTEM FUNCTIONAL CHECKOUT (CONT'D)

Step	Payload	Tech																																		
51.		BOB2 front panel Data sw - off Verify data LED - off Fan sw - off																																		
52.		Variable Power Supply Main pwr sw - off																																		
53.		Disconnect the following hardware: <ul style="list-style-type: none">Variable Power Supply from Power SourceBOB1 from Variable Power SupplyBOB2 from Variable Power Supply																																		
54.	Disassemble all Test Support System hardware Return the following hardware to Building 241 Controlled Storage: <table><thead><tr><th>Qty</th><th>Item</th></tr></thead><tbody><tr><td>1</td><td>TSS Laptop</td></tr><tr><td>2</td><td>DAQCARD</td></tr><tr><td>2</td><td>PA50-50F</td></tr><tr><td>2</td><td>DAQ 1200/SAB interface cable</td></tr><tr><td>1</td><td>SAB</td></tr><tr><td>1</td><td>Ethernet PCMCIA card</td></tr><tr><td>1</td><td>3C-PC-COMBO-CBL</td></tr><tr><td>1</td><td>Ethernet cable</td></tr><tr><td>1</td><td>Video Adapter cable</td></tr><tr><td>1</td><td>SAB/laptop Video cable</td></tr><tr><td>1</td><td>SAB/Laptop RS-232 cable</td></tr><tr><td>2</td><td>BOB</td></tr><tr><td>1</td><td>GSE Power cable</td></tr><tr><td>1</td><td>Variable Power Supply</td></tr><tr><td>1</td><td>Digital Multimeter</td></tr><tr><td>2</td><td>SAB/BOB Interface Cable</td></tr></tbody></table>	Qty	Item	1	TSS Laptop	2	DAQCARD	2	PA50-50F	2	DAQ 1200/SAB interface cable	1	SAB	1	Ethernet PCMCIA card	1	3C-PC-COMBO-CBL	1	Ethernet cable	1	Video Adapter cable	1	SAB/laptop Video cable	1	SAB/Laptop RS-232 cable	2	BOB	1	GSE Power cable	1	Variable Power Supply	1	Digital Multimeter	2	SAB/BOB Interface Cable	
Qty	Item																																			
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2	BOB																																			
1	GSE Power cable																																			
1	Variable Power Supply																																			
1	Digital Multimeter																																			
2	SAB/BOB Interface Cable																																			

T: _____ QA: _____

6.4 DEMATING OF PAYLOAD RACK FROM ISPR PANEL

6.4.1 Moderate Temperature Cooling Interface

6.4.1.1 PRCU Connector Savers

TABLE 6.33 PRCU CONNECTOR SAVERS

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	<p>Verify the following:</p> <ul style="list-style-type: none"> PRCU fluid sample is in spec per SSP30573 PEPSE facility breaker is OFF 	<p>Verify the following:</p> <ul style="list-style-type: none"> HRF Rack fluid sample is in spec per SSP30573 	
2.		<p>Inspect the following hardware for damage:</p> <ul style="list-style-type: none"> TCS Mod Supply hose CS Assy 683-27567-2 	
3.		<p>Inspect the following hardware for damage:</p> <ul style="list-style-type: none"> TCS Mod Return hose CS Assy 683-27567-4 	
4.		<p>Disconnect the following hardware:</p> <ul style="list-style-type: none"> TCS Mod Return hose P/N from CS Assy P/N 683-27567-4 	

TABLE 6.33 PRCU CONNECTOR SAVERS (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
5.		Disconnect the following hardware: <ul style="list-style-type: none"> CS Assy P/N 683-27567-4 from ISPR Panel (MOD TEMP TCS RETURN) 	
6.		Connect the following hardware: <ul style="list-style-type: none"> TCS Mod Supply hose bracket to T-bar Assy TCS Mod Return hose to T-bar Assy 	N/A: ____ T: ____ QA: ____

T: ____ QA: ____

6.4.1.2 GSE Transfer Hose

TABLE 6.34 GSE TRANSFER HOSE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> PRCU fluid sample is in spec per SSP30573 PEPSE facility breaker is OFF 	Verify the following: <ul style="list-style-type: none"> HRF Rack fluid sample is in spec per SSP30573 	
2.	Disconnect the following hardware: <ul style="list-style-type: none"> GSE Transfer Supply hose P/N from ISPR Panel (MOD TEMP TCS SUPPLY) 		
3.	Disconnect the following hardware: <ul style="list-style-type: none"> GSE Transfer Return hose P/N from ISPR Panel (MOD TEMP TCS RETURN) 		
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> GSE Transfer TCS Mod Supply hose GSE Transfer Return hose 	
5.		Inspect the following hardware for damage: <ul style="list-style-type: none"> TCS Mod Return hose 	
6.		Disconnect the following hardware: <ul style="list-style-type: none"> TCS Mod Supply hose from GSE Transfer Supply hose 	
7.		Disconnect the following hardware: <ul style="list-style-type: none"> TCS Mod Return hose from GSE Transfer Return hose 	N/A: ____ T: ____ QA: ____
8.		Disconnect the following hardware: <ul style="list-style-type: none"> GSE Transfer Supply hose to T-bar Assy GSE Transfer Return hose to T-bar Assy 	N/A: ____ T: ____ QA: ____

T: ____ QA: ____

6.4.2 GN₂ Interface

TABLE 6.35 GN₂ INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Disconnect the following hardware: • GN ₂ hose to CS Assy P/N 683-27587-1	
2.		Connect the following hardware: • CS Assy P/N 683-27587-1 from ISPR Panel (GN ₂)	
3.		Inspect the following hardware for damage: • GN ₂ hose • CS Assy 683-27587-1	
4.		Connect the following hardware: • GN ₂ hose to T-bar Assy	N/A: ____ T: ____ QA: ____

T: ____ QA: ____

6.4.3 Vacuum Interface

TABLE 6.36 VACUUM INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.		Disconnect the following hardware: • Vacuum Resource hose from CS Assy P/N 683-27552-1	
2.		Disconnect the following hardware: • CS Assy P/N 683-27552-1 from ISPR Panel (VRS)	
3.		Disconnect the following hardware: • Vacuum Waste Gas hose from CS Assy P/N 683-27552-2	
4.		Disconnect the following hardware: • CS Assy P/N 683-27552-2 from ISPR Panel (VES)	
5.		Inspect the following hardware for damage: • Vacuum Resource hose • CS Assy 683-27552-1	
6.		Inspect the following hardware for damage: • Vacuum Waste Gas hose • CS Assy 683-27552-2	
7.		Connect the following hardware: • Vacuum Waste Gas hose to T-bar Assy • Vacuum Resource hose to T-bar Assy	N/A: ____ T: ____ QA: ____

T: ____ QA: ____

6.4.4 Power Interfaces

6.4.4.1 Main Power Interface

TABLE 6.37 MAIN POWER INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none">POWER (J1) LED - offAUX POWER (J2) LED - off		
2.		Disconnect the following hardware: <ul style="list-style-type: none">UIP-P1 from CS Assy (J1) P/N 683-27524-1CS Assy (E1) P/N 683-27524-1 from Rack Handler Base	
3.	Disconnect the following hardware: <ul style="list-style-type: none">CS Assy (P1) P/N 683-27524-1 from ISPR Panel (J1 POWER)		
4.		Inspect the following hardware for damage: <ul style="list-style-type: none">UIP-P1 CableCS Assy P/N 683-27524-1	
5.	Connect the following hardware: <ul style="list-style-type: none">Dust cap to ISPR panel (J1)		
6.		Connect the following hardware: <ul style="list-style-type: none">UIP-P1 Cable to HRF Rack (Dust cap J1)	

T: _____ QA: _____

6.4.4.2 Auxiliary Power Interface

TABLE 6.38 AUXILIARY POWER INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none">POWER (J1) LED - offAUX POWER (J2) LED - off		
2.		Disconnect the following hardware: <ul style="list-style-type: none">UIP-P2 from CS Assy (J2) P/N 683-27524-2CS Assy (E1) P/N 683-27524-2 from Rack Handler Base	
3.	Disconnect the following hardware: <ul style="list-style-type: none">CS Assy (P2) P/N 683-27524-2 from ISPR Panel (J2 AUX POWER)		
4.		Inspect the following hardware for damage: <ul style="list-style-type: none">UIP-P2 CableCS Assy P/N 683-27524-2	
5.	Connect the following hardware: <ul style="list-style-type: none">Dust cap to ISPR panel (J2)		
6.		Connect the following hardware: <ul style="list-style-type: none">UIP-P2 Cable to HRF Rack (Dust cap J2)	

T: _____ QA: _____

6.4.5 Video Interface

TABLE 6.39 VIDEO INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> Main Power (J1) LED - off AUX Power (J2) LED - off 		
2.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P16 from CS Assy (J16) P/N 683-27524-6 	
3.	Disconnect the following hardware: <ul style="list-style-type: none"> CS Assy (P16) P/N 683-27524-6 from ISPR Panel (J16 VIDEO, LINE 1) CS Assy (P16) P/N 683-27524-6 from ISPR Panel (J16 VIDEO, LINE 2) CS Assy (P16) P/N 683-27524-6 from ISPR Panel (J16 VIDEO, SYNC) 		
4.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P77 from CS Assy (J77) P/N 683-27524-10 	N/A: ____ T: ____ QA: ____
5.		Disconnect the following hardware: <ul style="list-style-type: none"> CS Assy (P77) P/N 683-27524-10 from ISPR Panel (J77 VIDEO) 	N/A: ____ T: ____ QA: ____
6.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P77 Cable CS Assy P/N 683-27524-10 	N/A: ____ T: ____ QA: ____
7.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P16 Cable CS Assy P/N 683-27524-6 	
8.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P77 Cable to HRF Rack (Dust cap J77) 	N/A: ____ T: ____ QA: ____
9.	Connect the following hardware: <ul style="list-style-type: none"> Dust cap to ISPR panel (J16) 		
10.		Connect the following hardware: <ul style="list-style-type: none"> UIP-P16 Cable to HRF Rack (Dust cap J16) 	

T: ____ QA: ____

6.4.6 HRDL Interface

TABLE 6.40 HRDL INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> POWER (J1) LED - off AUX POWER (J2) LED - off 		
2.		Disconnect the following hardware: <ul style="list-style-type: none"> UIP-P7 from CS Assy (J7) P/N 683-27524-5 	
3.		Disconnect the following hardware: <ul style="list-style-type: none"> CS Assy (P1) P/N 683-27524-5 from ISPR Panel (J7-TX HRDL) 	
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> UIP-P7 Cable CS Assy P/N 683-27524-5 	

TABLE 6.40 HRDL INTERFACE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
5.	Connect the following hardware: • Dust cap to ISPR panel (J7)		
6.		Connect the following hardware: • UIP-P7 Cable to HRF Rack (Dust cap J7)	

T: _____ QA: _____

6.4.7 FDS Maintenance Interface

TABLE 6.41 FDS MAINTENANCE INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: • POWER (J1) LED - off • AUX POWER (J2) LED - off		
2.		Disconnect the following hardware: • UIP-P43 from CS Assy (J43) P/N 683-27524-7	
3.	Disconnect the following hardware: • CS Assy (P43) P/N 683-27524-7 from ISPR Panel (J43 FDS MAINT)		
4.		Inspect the following hardware for damage: • UIP-P43 Cable • CS Assy P/N 683-27524-7	
5.	Connect the following hardware: • Dust cap to ISPR panel (J43)		
6.		Connect the following hardware: • UIP-P43 Cable to HRF Rack (Dust cap J43)	

T: _____ QA: _____

6.4.8 1553B Interfaces6.4.8.1 Bus A Interface

TABLE 6.42 BUS A INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: • POWER (J1) LED - off • POWER (J2) LED - off		
2.		Disconnect the following hardware: • UIP-P3 from CS Assy (J3) P/N 683-27524-3	
3.	Disconnect the following hardware: • CS Assy (P3) P/N 683-27524-3 from ISPR Panel (J3 BUS A) • CS Assy (P7) P/N 683-27524-3 from 1553 ADDRESS TERMINATOR Assy (J7)		

TABLE 6.42 BUS A INTERFACE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P3 Cable • CS Assy P/N 683-27524-3 	
5.	Connect the following hardware: <ul style="list-style-type: none"> • Dust cap to ISPR panel (J3) 		
6.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P3 Cable to HRF Rack (Dust cap J3) 	

T: _____ QA: _____

6.4.8.2 Bus B Interface

TABLE 6.43 BUS B INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • POWER (J1) LED - off • POWER (J2) LED - off 		
2.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P4 from CS Assy (J4) P/N 683-27524-4 	
3.	Disconnect the following hardware: <ul style="list-style-type: none"> • CS Assy (P4) P/N 683-27524-4 from ISPR Panel (J4 BUS B) 		
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P4 Cable • CS Assy P/N 683-27524-4 	
5.	Connect the following hardware: <ul style="list-style-type: none"> • Dust cap to ISPR panel (J4) 		
6.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P4 Cable to HRF Rack (Dust cap J4) 	

T: _____ QA: _____

6.4.9 LAN Interfaces

6.4.9.1 LAN 1 Interface

TABLE 6.44 LAN 1 INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • POWER (J1) LED - off • POWER (J2) LED - off 		
2.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P46 from CS Assy (J46) P/N 683-27524-8 	
3.	Disconnect the following hardware: <ul style="list-style-type: none"> • CS Assy (P46) P/N 683-27524-8 from ISPR Panel (J46 LAN 1) 		

TABLE 6.44 LAN 1 INTERFACE (CONT'D)

Step	PRCU	HRF Rack 1	HRF Rack 2
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P46 Cable • CS Assy P/N 683-27524-8 	
5.	Connect the following hardware: <ul style="list-style-type: none"> • Dust cap to ISPR panel (J46) 		
6.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P46 Cable to HRF Rack (Dust cap J46) 	

T: _____ QA: _____

6.4.9.2 LAN 2 Interface

TABLE 6.45 LAN 2 INTERFACE

Step	PRCU	HRF Rack 1	HRF Rack 2
1.	Verify the following: <ul style="list-style-type: none"> • POWER (J1) LED is off • POWER (J2) LED is off 		
2.		Disconnect the following hardware: <ul style="list-style-type: none"> • UIP-P47 from CS Assy (J47) P/N 683-27524-9 	
3.	Disconnect the following hardware: <ul style="list-style-type: none"> • CS Assy (P47) P/N 683-27524-9 from ISPR Panel (J47 LAN 2) 		
4.		Inspect the following hardware for damage: <ul style="list-style-type: none"> • UIP-P47 Cable • CS Assy P/N 683-27524-9 	
5.	Connect the following hardware: <ul style="list-style-type: none"> • Dust cap to ISPR panel (J47) 		
6.		Connect the following hardware: <ul style="list-style-type: none"> • UIP-P47 Cable to HRF Rack (Dust cap J47) 	

T: _____ QA: _____

APPENDIX A

JSC Forms

These forms are for reference only.

		5. Page		of	
TASK PERFORMANCE SHEET CONTINUATION PAGE NASA - LYNDON B. JOHNSON SPACE CENTER		4. TPS NO.			
		6. MOD NO.			
20. OPER SEQ. NO.	21. OPERATIONS <i>(Print, Type, or Write Legibly)</i>			VERIFICATION	
				22. TECH.	23. QA/DV

JSC Form 1225A (Rev February 7, 2000) (MS Word August 1996)

Figure A-2 Task Performance Continuation Sheet

1. JPIC		Discrepancy Report/Material Review Record NASA - Lyndon B. Johnson Space Center				2. Page 1 of ____	
3. Ref. Doc. #		4. JCR #		5. DR #			
6. Name of Top Assy.		7. Drawing or P/N		8. S/N or Lot #		9. Qty.	
10. Name of Sub Assy		11. Drawing or P/N		12. S/N or Lot #		13. Qty.	
14. Name of Component		15. Drawing or P/N		16. S/N or Lot #		17. Qty.	
18. Description of nonconformance							
19. Initiator's name (print and sign)		20. Title/Stamp No.		21. Org.		22. Location	
						23. Date	
24. Responsible Engineer/Mail Code		25. CHRP Code		26. CAGE Code		27. Time/cycles used	
xx. Category		29. PRACA Reportable		30. Configuration Change?		31. Waiver?	
<input type="checkbox"/> Critical		<input type="checkbox"/> Yes <input type="checkbox"/> No		<input type="checkbox"/> Yes <input type="checkbox"/> No		<input type="checkbox"/> Yes <input type="checkbox"/> No	
<input type="checkbox"/> Major							
<input type="checkbox"/> Minor		FIAR # _____		DCN # _____		Waiver # _____	
						CAS # _____	
33. Final Disposition				34. MRR Ret'd?		35. Final Acceptance Stamp and Signature	
<input type="checkbox"/> Rework <input type="checkbox"/> Repair <input type="checkbox"/> Change Classification <input type="checkbox"/> Scrap <input type="checkbox"/> Use-as-is <input type="checkbox"/> Return to vendor/supplier <input type="checkbox"/> Written in error				<input type="checkbox"/> Yes <input type="checkbox"/> No			
Material Review Board <small>(Announcements must be typed or printed and signed)</small>							
36. Stress Engineer		Date		37. Materials Engineer		Date	
38. Project Engineer		Date		39. Quality Engineer		Date	
40. Other (print or type title)		Date		41. QA Rep. (NASA)		Date	
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recur. Ctrl.
							T9 Perf. Org.
							T10 Proc. Flow

JSC Form 2176 (Rev August 10, 1999) (MS Word Sep 97)

Figure A-3 Discrepancy Report/Material Review Record

1. IDR #	Discrepancy Report/Material Review Record	3. Page ____ of ____
2. DR #	NASA - Lyndon B. Johnson Space Center	
Continuation Sheet		
4. Insp. Pts.	5. Seq. No.	<div style="display: flex; justify-content: space-between;"> <div style="width: 60%;"> 6. Instructions <i>(Print, type, or write legibly)</i> </div> <div style="width: 35%;"> 7. Verification Stamps <div style="display: flex; justify-content: space-between; font-size: small;"> <div>Tech.</div> <div>Qual.</div> </div> </div> </div>
8. Final Acceptance Stamp and Date		
JSC Form 2176A (Sep 97) (MS Word Sep 97)		

Figure A-4 Discrepancy Report/Material Review Record Continuation Sheet

1. DR #	Discrepancy Report/Material Review Record NASA - Lyndon B. Johnson Space Center		2. Page ____ of ____
Summary Sheet			
3. Configuration Change? <input type="checkbox"/> No <input type="checkbox"/> Yes DCN #		4. CCBD #	5. PRACA #
6. Remedial Action			
7. Root Cause			
8. Corrective Action (Recurrence Control)			
MRB APPROVAL			
9. Stress Engineer (Print and sign)		10. Materials Engineer (Print and sign)	Date
11. Project Engineer (Print and sign)		12. Quality Engineer (Print and sign)	Date
13. Other (Print and sign)		14. QA Rep. (NASA) (Print and sign)	Date
JSC Form 2176B (Oct 97) (MS Word Sep 97)			

Figure A-5 Discrepancy Report/Material Review Record Summary Sheet

1. DR #	Discrepancy Report/Material Review Record NASA - Lyndon B. Johnson Space Center	2. Page ____ of ____							
Multiple Disposition Coding Sheet									
A.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
B.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
C.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
D.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
E.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
F.									
T1 Resp. Org.	T2 HW Type	T3 Prev. Cond.	T4 Fail. Mode	T5 Defect	T6 Remedial Act.	T7 Cause	T8 Recuf. Ctrl.	T9 Perf. Org.	T10 Proc. Flow
3. Quality Engineer (Print and Sign)						Date			
JSC Form 2176C (Oct 97) (MS Word Oct 97)									

Figure A-6 Discrepancy Report/Material Review Record Multiple
Disposition Coding Sheet



FLASH REPORT

For Safety and Product Assurance use only

NASA mishap no.	
OSHA file no.	
GENERAL INFORMATION	
1. Date (MM/DD/YY)	2. Time <input type="checkbox"/> a.m. or <input type="checkbox"/> p.m.
3. Building number/location	4. Specific area
5. Category of incident (check appropriate box)	
<input type="checkbox"/> Injury/accident <input type="checkbox"/> Fire <input type="checkbox"/> Auto accident <input type="checkbox"/> Explosion <input type="checkbox"/> Chemical spill <input type="checkbox"/> Other	
6. Description of incident (explain what happened, including cause or description of failure)	
7. SEAT involvement (name of organization)	
PERSONNEL INVOLVED	
8. Name (last, first, middle initial)	9. Telephone
CONTACT PERSON	
10. Name (last, first, middle initial)	11. Telephone

FORM SEAT 094 (09/23/97)

Figure A-7 Flash Report

[illegible]

Figure A-9 Repetitive Operations Log

PASS 1000 _____ V dc	MAIN	AUX
SSPCM Status - The SSPCM status bits/words show valid states.		
SSPCM BIT Status, (word 14)	Expected = x000, Diff = _____	Expected = x000, Diff = _____
SSPCM BIT Status, (word 15)	Expected = 0000, Diff = _____	Expected = 0000, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____
RIC Mode Status - The RIC is in Standby mode.		
MMC BIT and Status (word 260)	Expected = xxx1, Diff = _____	Expected = xxx1, Diff = _____
		TC: _____ Date: _____
RIC Card BIT Status - All RIC card status bits are valid.		
S1553C Bit and Status, (word 249)	Expected = 4000, Diff = _____	Expected = 4000, Diff = _____
HRLC Bit and Status, (word 253)	Expected = 0100, Diff = _____	Expected = 0100, Diff = _____
MCC Bit and Status, (word 258)	Expected = 0000, Diff = _____	Expected = 0000, Diff = _____
SERC BIT and Status, (word 261)	Expected = 0000, Diff = _____	Expected = 0000, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____
RIC Cards Heart Beat Status - The RIC card heart beat word values vary between each STEP update.		
MCC BIT and Status, (word 259)	Expected = Value Varies, Diff = _____	Expected = Value Varies, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____
PEHB Status - The PEHB status bits/words show valid states.		
PEHB POST Results, (word 148)	Expected = 0000, Diff = _____	Expected = 0000, Diff = _____
PEHB BIT Results, (word 149)	Expected = 0000, Diff = _____	Expected = 0000, Diff = _____
LAN 0 BIT and Status, (word 158)	Expected = 0010, Diff = _____	Expected = 0010, Diff = _____
LAN 0 BIT and Status, (word 159)	Expected = FFFF, Diff = _____	Expected = FFFF, Diff = _____
LAN 0 BIT and Status, (word 160)	Expected = 0003, Diff = _____	Expected = 0003, Diff = _____
LAN 1 BIT and Status, (word 213)	Expected = 0010, Diff = _____	Expected = 0010, Diff = _____
LAN 1 BIT and Status, (word 214)	Expected = 0003, Diff = _____	Expected = 0003, Diff = _____
LAN 2 BIT and Status, (word 231)	Expected = 0010, Diff = _____	Expected = 0010, Diff = _____
LAN 2 BIT and Status, (word 232)	Expected = 0003, Diff = _____	Expected = 0003, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____
HRF Rack Equipment Channel Configurations - Verify, on the STEP "Health & Status Data Display" screen, that the rack channels are configured.		
Channel 1(DDC)/Channel 0(DC to DC Converter(DDC)) (word 133, MSB/LSB)	Expected = 3C3C, Diff = _____	Expected = 3C3C, Diff = _____
Channel 3(Spare)/Channel 2(RIC) (word 134, MSB/LSB)	Expected = 183C, Diff = _____	Expected = 183C, Diff = _____
Channel 5(Spare)/Channel 4(PEHB) (word 135, MSB/LSB)	Expected = 183C, Diff = _____	Expected = 183C, Diff = _____
Channel 7(DDC)/Channel 6(FDS) (word 136, MSB/LSB)	Expected = 3C3C, Diff = _____	Expected = 3C3C, Diff = _____
Channel 9(EMU)/Channel 8(LAP) (word 137, MSB/LSB)	Expected = 3C3C, Diff = _____	Expected = 3C3C, Diff = _____
Channel 11(Valve)/Channel 10(Valve) (word 138, MSB/LSB)	Expected = 3C3C, Diff = _____	Expected = 3C3C, Diff = _____
Channel 13(HRF Fan)/Channel 12(Valve) (word 139, MSB/LSB)	Expected = 3C3C, Diff = _____	Expected = 3C3C, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____
Verify, on the STEP "Health & Status Data Display" screen, that the payload channels are configured.		
Channel 15(SIR Drawer Location 2)/Channel 14(SD1) (word 140, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 17(SD4)/Channel 16(SD3) (word 141, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 19(SD6)/Channel 18(SD5) (word 142, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 21(SD8)/Channel 20(SD7) (word 143, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 23(SD10)/Channel 22(SD9) (word 144, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 25(SD12)/Channel 24(SD11) (word 145, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 27(SD14)/Channel 26(SD13) (word 146, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
Channel 29(FP)/Channel 28(SD15) (word 147, MSB/LSB)	Expected = 1818, Diff = _____	Expected = 1818, Diff = _____
	TC: _____ Date: _____	TC: _____ Date: _____

Figure A-10 Health and Status Data Sheet

Deviation							Page ____ of ____
TPS Number:			Document Number:		Project Manager:		Test Engineer:
Dev No	Section	Step	Type (P/T)	Change		Reason	
Originator:			Phone:		Date:		Quality Engineer:

Figure A-11 Deviation Sheet

Deviation Continuation Page				TPS Number:	Document Number:	Page ____ of ____
Dev No	Section	Step	Type (P/T)	Change	Reason	

Figure A-12 Deviation Continuation Sheet

APPENDIX B

Illustrations

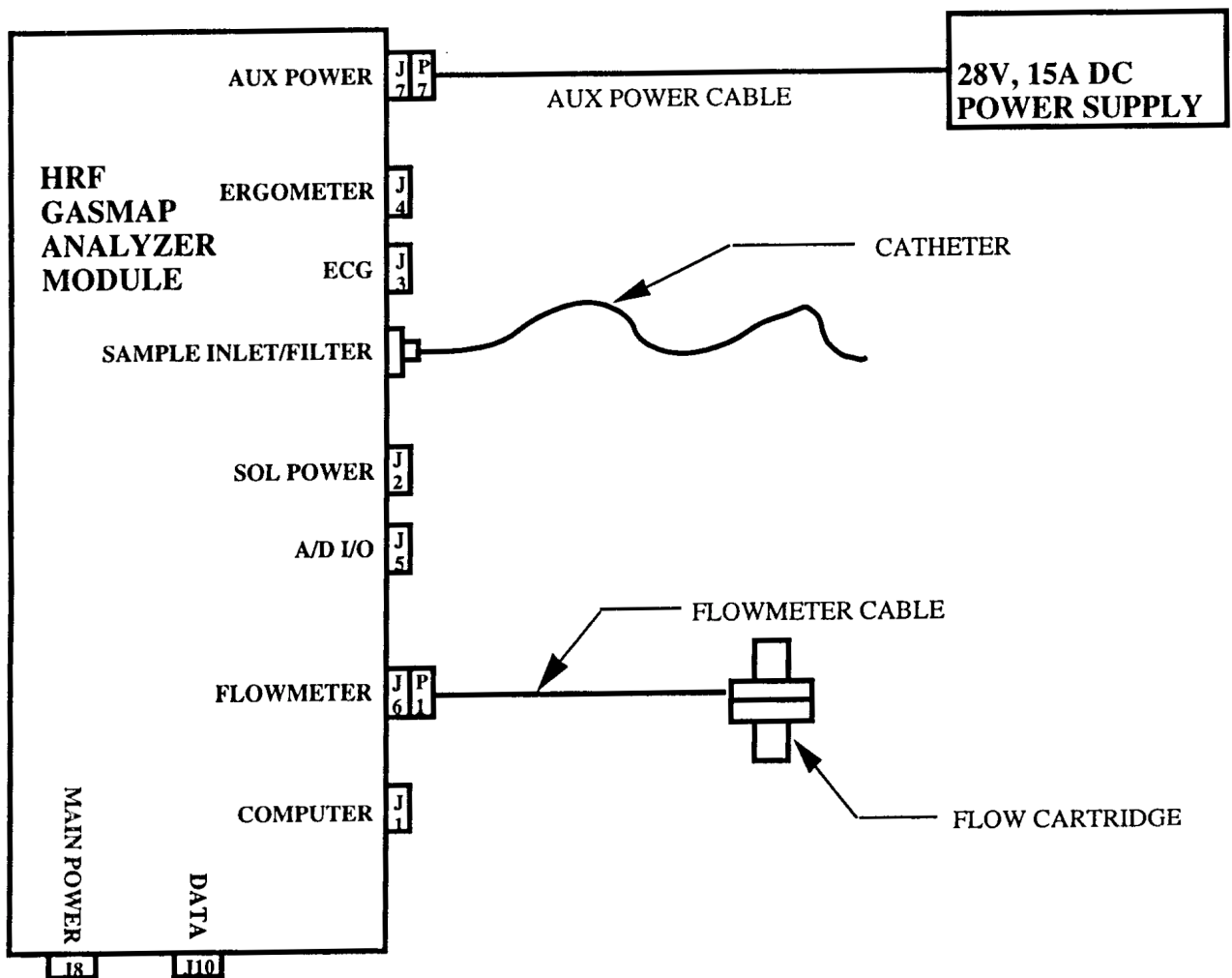


Figure B-1 GASMAP Assembly

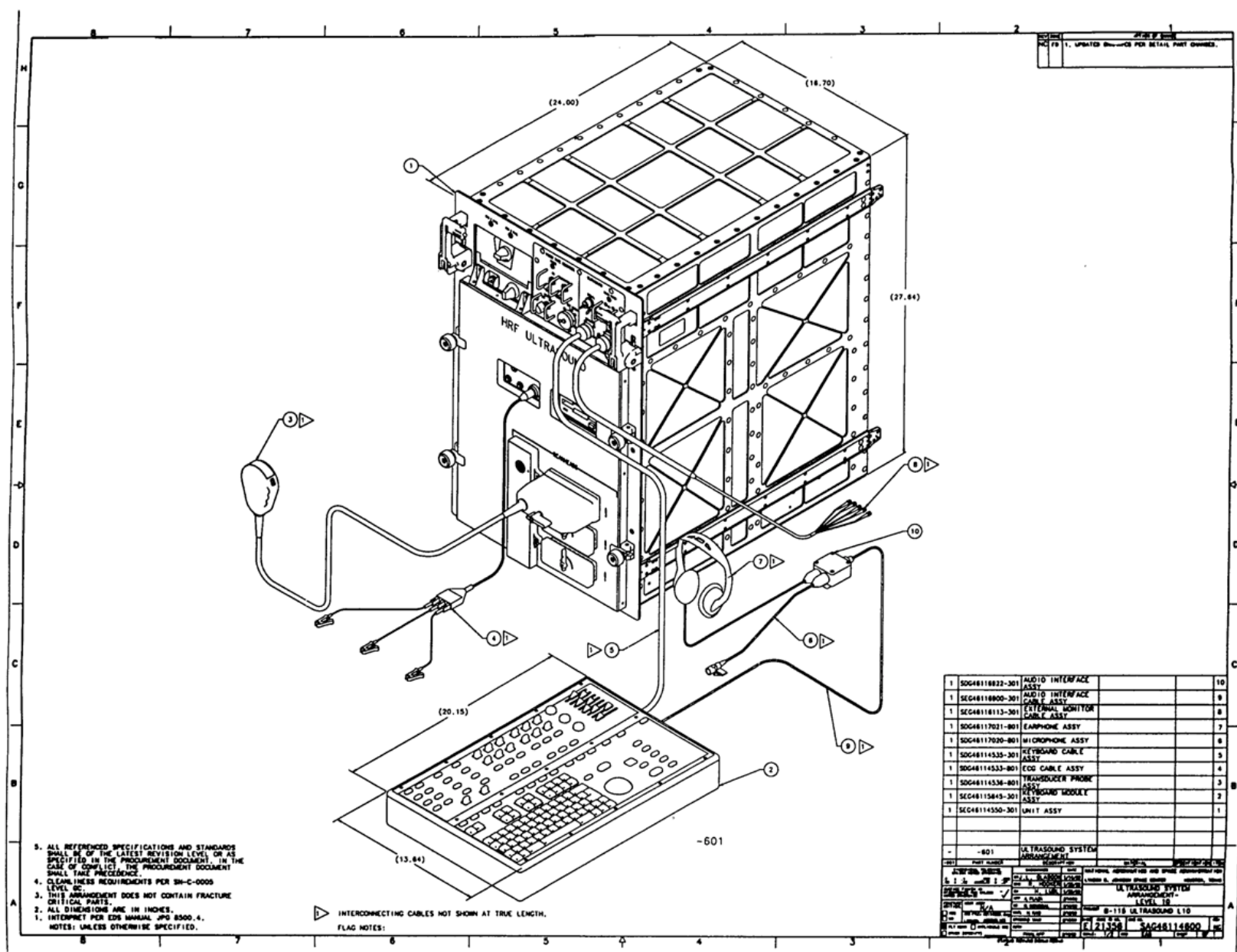
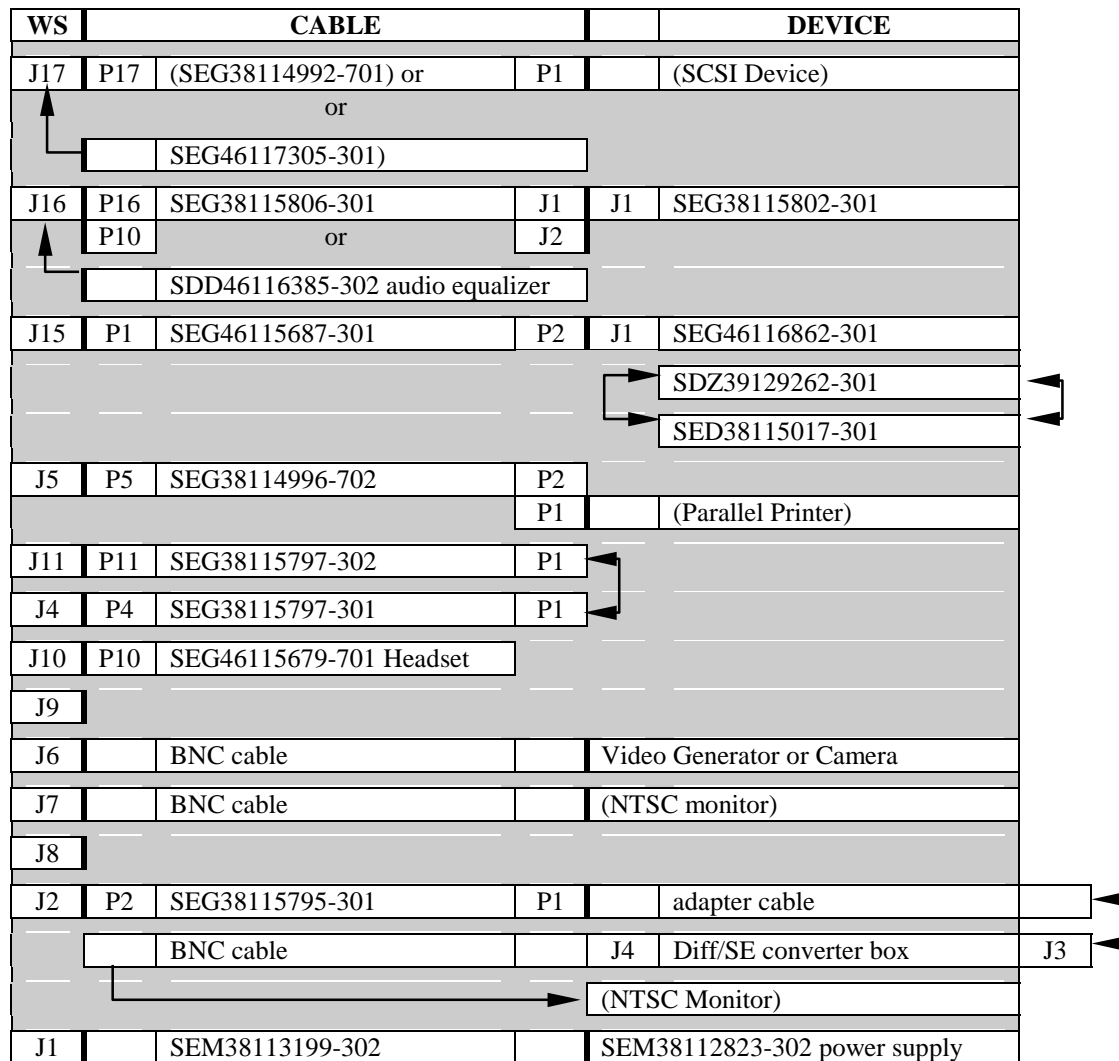


Figure B-2 Ultrasound Assembly



NOTE: Items in parentheses () are optional. They may be used but are not always required or are not always connected.

Figure B-3 Connections for WS Testing

WS	CABLE			DEVICE	
J14	P14	SEG38115015-301	P1		(Monitor)
			P2		Keyboard
			P3		
J12	P12	SEG38115796-301	P2		Mouse
			P3	P1	SEG38115003-701
			P4	P1	SEG38115003-701
			P5		
J3	P3	SEG38114993-303	P1		(Monitor)
			P2		(Monitor)
			P4		
			P5		

Figure B-4 Non-Flight Monitor and/or Keyboard Connections to WS

WS	CABLE			DEVICE	
28V PW R	P1	SEG46115683-301	P2	pwr	FP1610HB/R-06 Monitor
			P1		SEG46115492-301 Cable
			P2		
J14	P14	SEG46115494-301	P1		SEG46114997-301 Keyboard
			P2		
J12	P1	SEG46115686-302	P2		
			P3		P5 loopback
			P4		P6 loopback
J3	P3	SEG46115490-301	J1		
			J2		

NOTE 1: Dotted lines indicate alternative connections.

NOTE 2: Serial Breakout Cable SEG46115686-302 has as its cap a mating connector wired for loopback function. The Ground Support Equipment (GSE) Loopback Assembly SEG38115003-701 is neither needed nor compatible with this cable.

Figure B-5 Flight Monitor and/or Keyboard Connections to WS

WS	CABLE			DEVICE	
J11	P11	SEG46115489-301	P2		
J4	P4		P5		

Figure B-6 Connections for WS Testing Using Other Flight Items

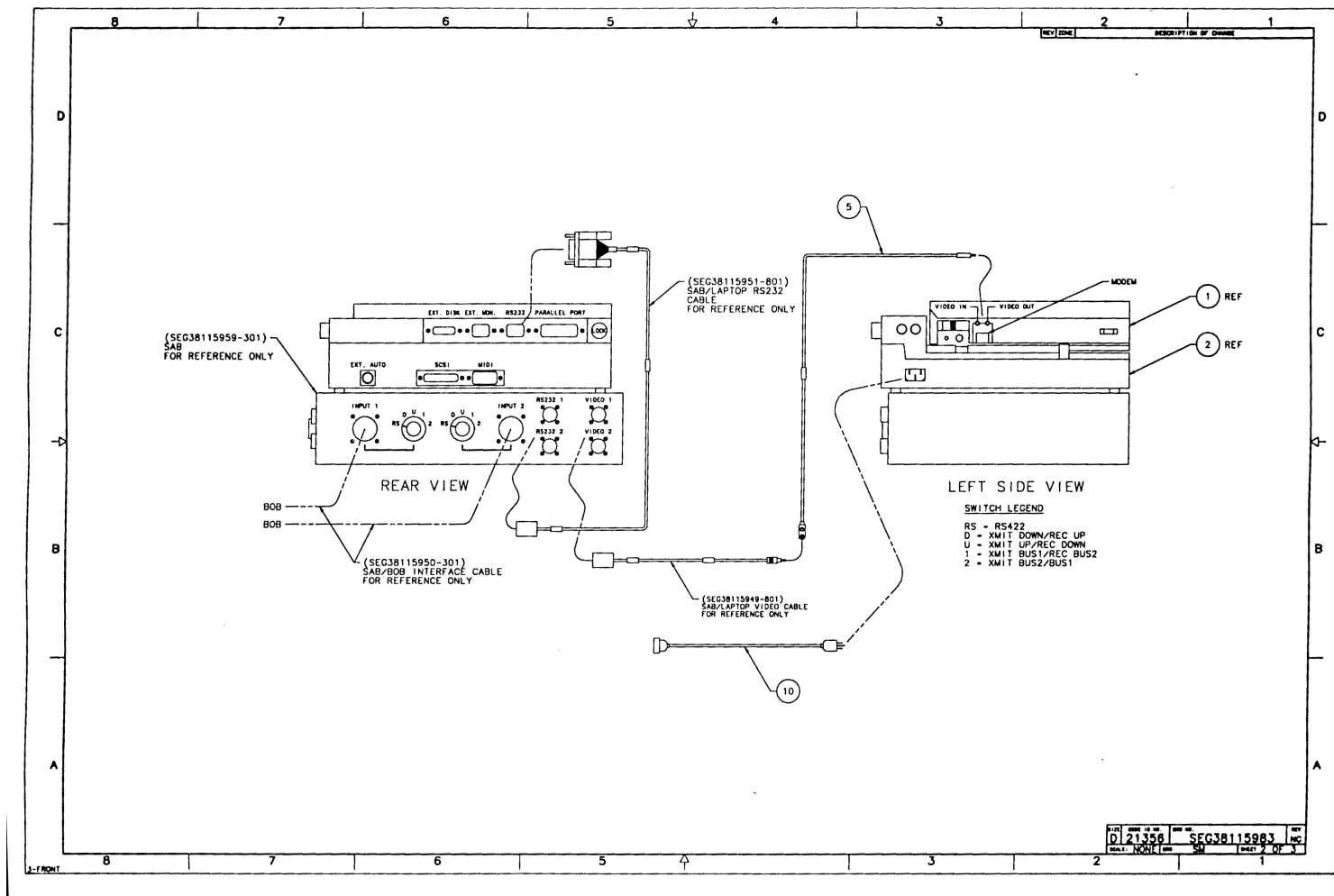


Figure B-7 Test Support System Assembly Rear View and Left Side View

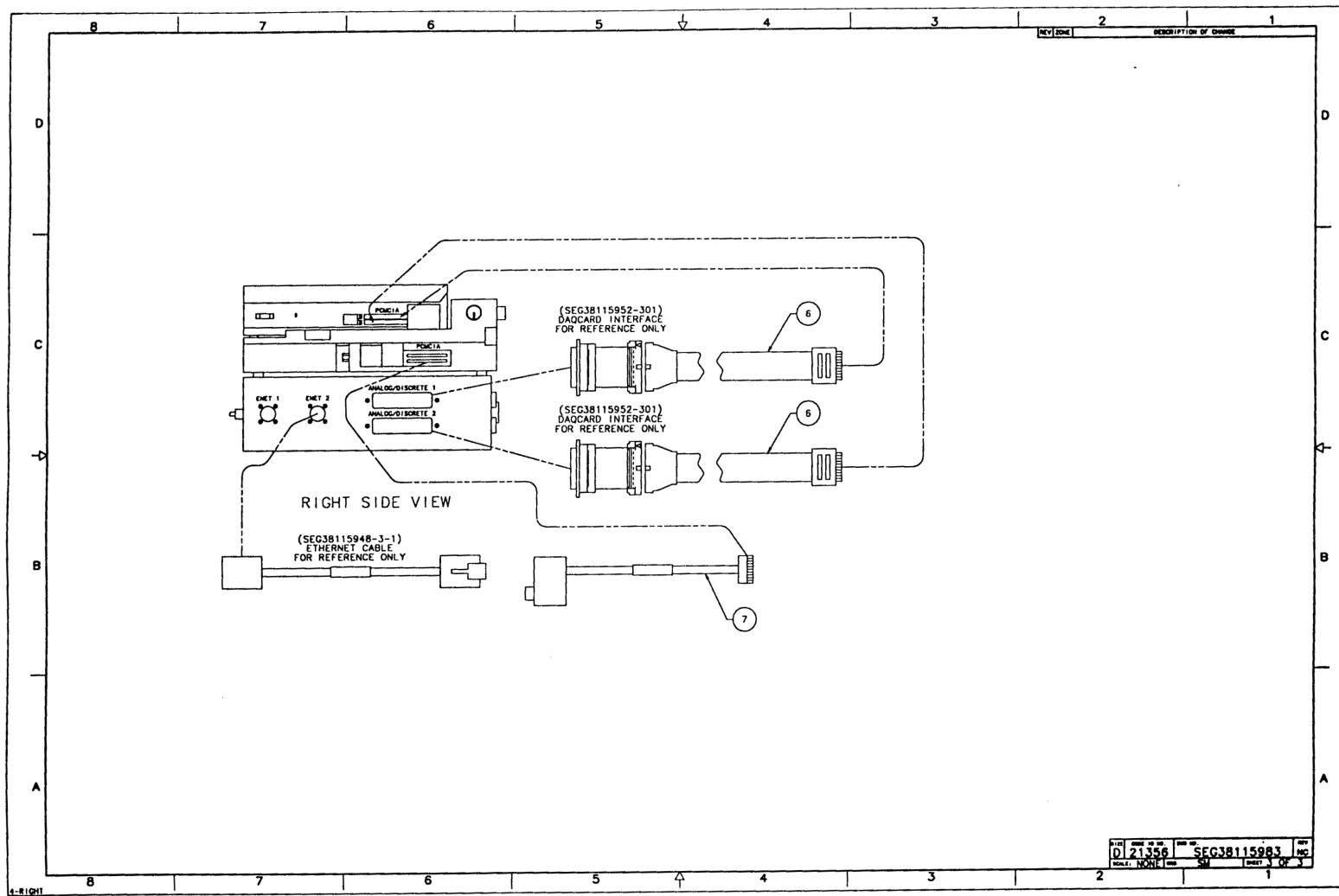


Figure B-8 Test Support System Assembly Right Side View

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